

Aviation Week & Space Technology

October 21, 1963

Sweden Plans
1970s Air Force
Around Viggen

New Centaur Vent Stack
Tested at Lewis Center

75 Cents

A McGraw-Hill Publication



materials / 63/64



HANDBOOK OF FASTENER TECHNOLOGY



STRENGTH OF MATERIALS

HIGH TENSILE NUTS

WATER-RESISTANT

High-Temperature Bolts

METALLURGY

Volumes of Performance from Voi-Shan

Material	Room Temperature Minimum Ultimate Tensile Strength (Tensile)	Typical Temperature Range	Typical Head Configurations
A308 Steel A 286 Corrosion Resistant Steel B6 Ti-62	150,000 psi 150,000 psi 170,000 psi	-100 thru +500°F -423 thru +1700°F -423 thru +1450°F	100° Flat Head, Internal or External Hex, or Extended 12 pt. Wrenching (Type 6, MIL-S-27462 Class 3A)
A308 Steel B6 Ti-62 A 286 Corrosion Resistant Steel B6 Ti-62	180,000 psi 200,000 psi 240,000 psi	-100 thru +500°F -423 thru +1800°F -423 thru +1600°F	100° Flat Head, Internal or External Hex, or Extended 12 pt. Wrenching
A 286 Corrosion Resistant Steel Ti-6Al-4V	200,000 psi 200,000 psi	-423 thru +1700°F -325 thru +750°F	Threads, MIL-S-27462 B6, S 8579 Class 3A
Hy-Tuf Vermet SiC, Chrom SiC, Chrom SiC, Chrom	220,000 psi 220,000 psi 260,000 psi 260,000 psi	-100 thru +1800°F -100 thru +1800°F -100 thru +1800°F -100 thru +1800°F	Internal or External Hex or Extended 12 pt. Wrenching
Inconel 718 Inconel 718 Inconel 718	260,000 psi 260,000 psi 300,000 psi	-423 thru +1800°F -423 thru +1800°F -423 thru +1800°F	Threads, MIL-S-27462 Class 3A

See also a type

REFRACTORY FASTENERS

Wolfram
Columbium
Tantalum
Tungsten

Minimum ultimate tensile
strength varies according to
composition, grain size, and
application temperature

up to +3500°F
up to +2600°F
up to +2600°F
up to +1800°F

Flash and Preheated
Head Styles
Threads, Voi-Shan
Special Design



on target with turboshaft power for the LOH—the Light Observation Helicopter which will soon be helping the Army fulfill its mission. The only LOH engine to have both US Army approval and FAA certification, the Allison T63 was selected for the LOH because of its light weight, power, simplicity, compactness and ease of maintenance. Now proving its capabilities in airframe evaluation flight tests for the Army, the T63 has over 1,000 flight hours, hit every performance and delivery target set for it. That kind of bull's eye performance is the big reason why our other aerospace and nuclear programs are also on target.

Allison
THE TURBO COMBUSTION SPECIALISTS OF
GENERAL MOTORS ENGINE DIVISION



FIRST AS A MATTER OF RECORD... SCOTCH® BRAND INSTRUMENTATION TAPE.



The tape with the built-in duster!

1000 times greater conductivity than ordinary tapes! That's how "Scotch" brand Heavy Duty Tapes don't off static charges before they can attract dust. That's the built-in duster that kicks away the growing danger of dust-caused equipment errors... a danger greater than ever as higher and higher recorder speeds and tape tensions generate more and more static.

Electrical resistance of the music coating of "Scotch" Heavy Duty tapes is 30 megohms per square or less. The insulating conductivity, instead of insulating tape, not only erases their static problems, it eliminates their other static problems as tape drag and skew, noise induced by noise.

Heavy duty formulation of binder and heat potency enables Scotch tapes to operate at temperatures from -40 to as high as +258°F to conquer high heat!

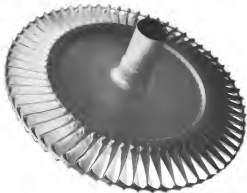


here, across tapes that outlast standard tapes by at least 15 times. Exclusive Silicone Lubrication reduces recorder head and tape wear. And "Scotch" Heavy Duty Tapes are offered for all high-speed applications, even for extreme high frequency and short wavelength requirements. 15 different constructions include a variety of backing and enable thicknesses.

TECHNICAL TALK Bulletin No. 4 provides detailed discussion of the effects of some electrolytic instrumentation recordings, offers helpful information on solving static-caused problems. It's free. Write: 3M Magnetic Products Division, Dept. ME2-333, St. Paul 78, Minn.

*TYPICAL. SEE OUR CATALOG FOR COMPLETE DATA. THIS TAPE IS AVAILABLE IN 1/4, 1/2, 3/4, 1, 2, 3, 4, 5, 6, 8, 10, 12, 15, 20, 25, 30, 35, 40, 45, 50, 60, 70, 80, 90, 100, 120, 150, 180, 200, 250, 300, 350, 400, 450, 500, 600, 700, 800, 900, 1000, 1200, 1500, 1800, 2000, 2500, 3000, 3500, 4000, 4500, 5000, 6000, 7000, 8000, 9000, 10000, 12000, 15000, 18000, 20000, 25000, 30000, 35000, 40000, 45000, 50000, 60000, 70000, 80000, 90000, 100000, 120000, 150000, 180000, 200000, 250000, 300000, 350000, 400000, 450000, 500000, 600000, 700000, 800000, 900000, 1000000, 1200000, 1500000, 1800000, 2000000, 2500000, 3000000, 3500000, 4000000, 4500000, 5000000, 6000000, 7000000, 8000000, 9000000, 10000000, 12000000, 15000000, 18000000, 20000000, 25000000, 30000000, 35000000, 40000000, 45000000, 50000000, 60000000, 70000000, 80000000, 90000000, 100000000, 120000000, 150000000, 180000000, 200000000, 250000000, 300000000, 350000000, 400000000, 450000000, 500000000, 600000000, 700000000, 800000000, 900000000, 1000000000, 1200000000, 1500000000, 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HERE'S A LOW-ALLOY, LOW-COST TIMKEN® STEEL THAT MADE THE GRADE IN FLIGHT



Annealinging a 17-22-A steel engine gear for use with 17-22-A steel containing less than 2% alloy

IT CAN KEEP YOUR COSTS DOWN TO EARTH

Although it has less than 2% alloy content, Timken® "17-22-A" steel holds its maximum strength up to 1180°F. It has a further advantage over many exotic high alloy steels in hot working. You can forge or work this steel up to 2300°F—machine and weld it easily. Resistance to heat checking and thermal cracking is excellent. Normalizing and tempering develop the maximum high temperature characteristics of "17-22-A" steel. And there's almost no possibility of distortion and quench cracking.

Timken "17-22-A" steel has been widely used for

aircraft brake discs for years, and now is being used more and more in gas turbine engine discs, as well as other aircraft and space applications. If you want more information about "17-22-A", send for our free 44-page Technical Bulletin 365, The Timken Roller Bearing Company, Steel and Tube Division, Canton 6, Ohio. Makers of Tapered Roller Bearings, Fine Alloy Steel and Removable Rock Bits.

TIMKEN® FINE ALLOY STEEL

TIMKEN ALLOY STEEL, AND STAINLESS STEEL, TUBES AND ANNUALLY FROM STEEL SERVICE CENTERS IN MORE THAN 70 CITIES IN THE UNITED STATES



What you can do with General Electric's versatile **RTV silicone compounds** to insulate, seal and mold from -150°F to 500°F



Example 1: Fluid RTV silicone rubber penetrates deep into inaccessible cavity. RTV has excellent dielectric strength and practically no shrinkage. Cure time at room temperature can be varied from minutes to hours.



Example 2: Tempering as required, GE silicones provide a excellent protection against corrosion, erosion, chemical and mechanical attack. They're easily removed from complicated parts, run by cut away to replace internal components.



Example 3: Flexible RTV is often used to make molds for prototypes and short-run production. This part requires very uniform, long-life dielectric parts. Cure is readily, RTV's tensile strength is as high as 500 psi.



Example 4: Dielectric RTV (white and black) is generally poured into open and machine and clean. Some can be dissolved. Can be used for sheet metal fabrication, sheet metal, painting. Visitation range from possible to paste.



Example 5: Adhesive/sealant RTV-900 is applied as a bonding agent. Can be used to provide open wiring for on demand cooling, plating and soldering. RTVs are usually applied, will not shrink, crack or weather.



Example 6: RTV adhesive/sealants are hot working correctly made, clean, pre-fabricated parts or more easily use existing techniques. Here an RTV adhesive/sealant. Flexible metal strips in form cylindrical shells.

If you would like a free sample of any of the nine General Electric RTV silicones for evaluation, write us your letterhead describing your application. For additional information, check reader service card. Section 1104, Science Products Dept., General Electric Company, Westfield, New York.

GENERAL ELECTRIC

RESEARCH

to fire the imagination, advance the state of the art

Research at THIOKOL covers the whole spectrum of Science

There is no discipline in the physical sciences which Thiokol does not so actively investigate. While THIOKOL's major interest is the utilization of research to improve rocket engine performance and reliability—this work is based on the most fundamental concepts of science. Maxwell's electromagnetic equations, for example



James Maxwell laid these foundation blocks of electro-magnetic theory in 1864. Then, the only practical source of electricity was the chemical battery, and its only widespread use, the telegraph. Today, Thiokol is applying Maxwell's Equations to the relationships between electromagnetic fields and plasmas. Spectroscopic, microwave and other methods are used to inhibit temperature, charge concentrations and other plasma properties to prevent their. Applications of interest are magnetic hydrodynamic thrust and power generation, and propagation of radio frequency spectrum through weakly ionized gas

Structure of Macromolecules

In the realm of the investigation of the composition and structure of molecules, Thiokol Chemical Corporation has left the test tube behind. The Nuclear Magnetic Resonance Spectrometer shown here is one of



the new tools of science we are utilizing in this area. This device takes advantage of the fact that certain atomic nuclei have magnetic moments. When a compound containing such nuclei is placed in a magnetic field and subjected to radio frequency energy, oscillations occur which can be recorded. By this means a large variety of groups and bonds can be differentiated. This technique, in conjunction with other modern chemical analytical tools such as infra red, mass spectrometer, ultra violet absorption, gas chromatography will determine rapidly and efficiently the structure of the most intricate molecule. Samples as small as a milligram of a pure can be analyzed and, quite often, need not be destroyed by analysis. With these analytical tools, Thiokol is studying the structure of the complex macromolecules that find use in such diverse areas as pro-

pelant binders, potting compounds and structural-joint sealants.



Anatomy of Flames

Flames are the most complex high temperature chemical systems known. The above structured rocket exhaust, illustrating the conversion of random heat motion to directed thrust, is but one of their many applications. At Thiokol we are probing flames and high velocity exhausts for chemical species, as well as determining macroscopic properties. Light emission and absorption, from the ultra-violet to the infrared are also measured. These results provide a basis for the development of more fuels, for the design of more efficient rocket components and for the determination of the properties of missile exhaust.

Reservoir of knowledge

Thiokol research in its past and applied forms is adding to the total sum of man's knowledge. Challenging experiments exist for creative scientists and interested in these areas. Applications are welcome.

Thiokol
CHEMICAL CORPORATION
Route 1, Peabody, Mass.
Rocket Division Center, Oyster, Utah
An equal opportunity employer

FOR HELICOPTERS:

new ideas in vibration shock/force control



Longer service.

Less bounce.

More stability.

Helicopter design has made much progress in solving vibration problems, improving comfort, reducing structural fatigue. ■ Here are three recent contributions from Lord for helicopters. **Longer service** results from a special "soft" drive coupling which transmits power smoothly, reduces vibrating torques in main transmissions, extends overhaul periods. **Less bounce** is the feature of a landing gear stabilizer. It cushions landing gear shocks and controls ground resonance effects. **More stability** is achieved by a special damper which controls helicopter rocking motions. It is used in a Lord low-frequency rotor suspension system. ■ These are only three of many Lord developments to improve helicopter performance. Talk to us first on your problem—then expect more. Contact: Lord Manufacturing Company, Erie, Pa. Field Engineering Offices in principal cities. In Canada: Railway & Power Engineering Corp., Ltd.





**From Fafnir -
computer-programmed
bearing reliability
for turbine-powered
helicopters**

There was when helicopter transmission bearings could be selected right from the catalogs. But no more. In turbine-powered 'copters, especially, many gearbox bearings must be custom-engineered.

Fafnir is a major supplier of such bearings — and the leader in applying advanced computer techniques to bearing engineering. The gearbox bearings shown above are good examples.

The application demanded extremely rigid shaft support and high running accuracies under conditions of severe vibration. Bearing load capacity, weight, and size were critical, too.

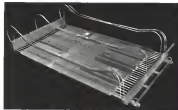
An IBM 7090 computer analyses helped Fafnir engineers explore design alternatives, and determine optimum bearing characteristics. Result: a duplex set of angular contact bearings precisely engineered for maximum performance and load-life requirements.

Chances are, the "computer approach" at Fafnir can help you meet critical needs precisely . . . and with utmost reliability. Investigate. Remember "Fafnir" on your bearings assures sound engineering, highest quality, full value at a fair price. The Fafnir Bearing Company, New Britain, Conn.

FAFNIR
BALL BEARINGS

**FAFNIR DUPLEX
BEARING SET FOR
HELICOPTER GEARBOX**

Flanged M type bearing with companion M2 type bearing (manufactured to ASCE) flange. Rings and balls of high-strength vacuum-heat-treated steel made by consumable electrode process are 100% magnetic particle and surface temper inspected. Machined bronze retainers are 100% fluorescent penetrant inspected.



Easy-B Mounting Construction



With External Sealing PG 1010



Thin Oil Grooves For Film PG 1010



Machine Bearings Design PG 1010



With External Sealing PG 1010



Sealing Oil Resistant Lubrication PG 1010



Sealing Oil Resistant Lubrication PG 1010



Machine Bearings Design PG 1010

INFINITY

Fins, fins, fins. Plain fins, slotted fins, sullied fins, laminated fins, stepped fins. Cold and hot fluid create fins of different heights, thicknesses, densities and materials. UAP can design and build secondary surface heat exchangers of the plate-and-fin type to meet your precise requirements. We make all types of fins in our own plant, and employ our own outstanding dip-braze facilities. This shortens lead time, gives you better delivery. And so speed design time, UAP has programmed a digital computer for an infinite amount of heat exchanger information derived from experience and testing, plus engineering performance data. The computer can quickly indicate what type of fin is best for your application in terms

of pressure drop, heat rejection, weight, size and other factors. For details, write . . . or phone 224-3441 today. UAP means United Airmat Products. Since 1929 a dynamic, independent company in Dayton, Ohio. A name to remember when it comes to plate-and-fin heat exchangers.

Outstanding opportunities for qualified engineers.





Space- the youthful outlook

Space-General offers unlimited opportunities for engineers and scientists who want to move ahead quickly in a dynamic young organization dedicated to space projects. ■ Less than three years old, Space-General now has grown to almost 1,800 employees and occupies one of the newest, best-equipped facilities in the industry. Location is ideal...in the San Gabriel Valley, 15 minutes by freeway from downtown Los Angeles. ■ Our current interests are broad: electronics, guidance and control systems, space telemetry, survivable communications, satellite and military systems, space vehicles, laser and planetary exploration craft, inflatable ramjet vehicles, and nuclear defense studies. ■ Our employees tell us they prize Space-General's unusually heavy emphasis on individual work and ideas. As a matter of fact, every member of our technical staff is expected to make a decisive contribution to the growth of our technological capability.



SPACE-GENERAL CORPORATION

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There are many outstanding career opportunities at Space-General for engineers and scientists with a B.S. degree or higher and at least two years of experience in any of the following fields: Communication Systems Analysis • Circuit Design • Heat Transfer Thermodynamics Aerodynamics • Stress Analysis • Trajectory Analysis and Aerodynamics • Hydrodynamic Control Development • Structural Test Analysis • Space Vehicle Design • Guidance and Control Analysis • Advanced Sensor Research • Micro-wave Research • Nuclear Test Instrumentation • Bio-Sensor Research. ■ Your resume will receive immediate confidential attention. Send to: Donald L. Craig, Employment Mgr., Space-General Corp., Dept. No. NP 15, 2000 E. Flair Drive, El Monte, California.

An equal opportunity employer.



DALMO VICTOR PUTS "SATAN" TO WORK FOR NASA Project "Satan" improves capability and reliability for data acquisition, telemetry and command. Thus NASA's Goddard Space Flight Center's worldwide Satellite Tracking and Data Acquisition stations perform pre-designed tasks on the various satellite programs—automatically and with minimum maintenance. □ Project "Satan" is an example of the key role played by Dalmo Victor in development work with NASA schedules, and in the manufacture of high-performance aerospace systems. DV is in the vanguard of new developments in its major product areas. If you are interested in becoming part of these challenging programs, Dalmo Victor currently is inviting applications from qualified scientists and engineers. For further information contact: Director, Scientific and Engineering Personnel.

DALMO VICTOR COMPANY 1201 REDWOODS BOULEVARD • DELMAR, CALIF.



RAYTHEON

reflex KLYSTRONS for missile environments

Regulated parts... specially designed grids to eliminate unwanted low-voltage electronic tuning... high thermal stability and minimize warm-up frequency drift. These are a few of the features that put Raytheon reflex klystrons in a class by themselves for reliable operation in missile environments.

Designed for parametric amplifier and local oscillator service at 16 to 40 KMC, Raytheon reflex klystrons also feature lightweight, compact construction (see photo for actual size). They are available with power output ratings ranging from 10 to 200 mW.

Get in touch with your Raytheon Sales Engineer for all the facts and figures on reliable Raytheon-reflex klystrons for missiles. Or, write Raytheon Company, Microwave and Power Tube Division, Waltham, Massachusetts 02154.



RAYTHEON

Photo: Tom Arnold/Esso

Volume 39
Number 17

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October 31, 1963

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EDITORIAL

Racing Again

COVER: Scale model of a Conquest launch vehicle with a redesigned liquid hydrogen vent stack (shown in background) undergoing a wind tunnel test at Neilson Aerodynamics and Space Laboratories' Lewis Research Center. Engineers want to keep the hydrogen plume away from the crew capsule to avoid ignition. For more information on Conquest, see p. 15.

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FEATURES

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Member AEP and AEC

11,632 copies of this issue printed

Racing Again

After some sad intervals of indecision, the United States last week apparently got back into two of the key technological races of the decade with a renewed determination to try to win them. The boost gives the super-space transport program by the willingness of four U.S. senators to vote down payments for delivery positions on a theoretical production line should certainly stimulate the bureaucrats who are running that program to more vigorous effort. It should also indicate to Congress that this is an important national effort with a solid economic price at stake in addition to whatever prestige may be involved.

It is somehow typical of the manner in which any federal program seems bound to operate that, when confronted with certified checks totalling some \$2.1 million, the Federal Aviation Agency was paralyzed because it could not find the correct bureaucratic slot in which to deposit them. We rather doubt if our Anglo-French competitors would display FAA Administrator Noyah Hahay's reluctance to accept such quantities of cash.

Canny Managers

There is a much vagueness about these orders as there is about what kind of a super-space transport will eventually be the U.S. flag. Certainly such canny airline managers as C. B. Smith of American and Juan Trippe of Pan American are not going to stake their competitive fortunes solely on the FAA's eventual product or the pace of congressional support for it. They have wisely hedged by taking places on the Anglo-French Concorde production line. But their early willingness to let several million dollars on the U.S. entry in the super-space competition has served an extremely useful purpose by making it clear to Congress, the President and key executive ministers in the FAA that the proposed product itself has a viable market and significant place in the future of air transport.

Although it has a much lower development price tag, the technical and managerial problems involved in bringing a super-space transport into successful airline service are nearly as complex and difficult as those facing the Apollo lunar landing program. It would be a great mistake to underestimate the difficulties facing the super-space transport program. It would be a mistake of equal magnitude to let these difficulties divert this country from tackling the job with all of the technical, managerial and fiscal resources it can marshal.

President Kennedy feared his proposal for a joint U.S.-USSR lunar landing program about equally unpopular in both countries, and apparently was surprised by the

vehementness of his domestic critics of the project. The Russians also made it clear that they are soiled (owing to the Moon, that dislodging some of the fog generated by the sage of Jeddah) Bank, Sir Bernard Lovell, that temporarily convinced some people in high places that the Soviets lacked both the capability and intent to land men on the moon. Since the Soviets feared the large side telescope over which he presides useful in tracking their deep space probes, they have also found an additional use for Sir Bernard as an international disseminator of their overblowing propaganda lines.

Executive Irresponsibility

Although much of the legislative damage wrought by the President's hastily conceived proposal has been repaired by frantic party whip-cranking, most of the space program's staunchest supporters are still checked by this display of executive irresponsibility toward one of the key programs on which the incumbent Administration must face the elections next fall.

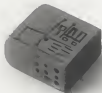
The key management of NASA has apparently learned little from its heavy rebuffs by Congress on the current appropriations bill and shows no inclination to put its non-essential activities to save the critical funds for its more technical programs.

However, mingled with satisfaction that both of these key programs are again gaining momentum and support is a disappointment that this country did not initiate either venture, but was pushed into them somewhat reluctantly by the type of foreign competition. It is this U.S. unwillingness to take the lead in exploring bold new frontiers that is at the root of many of its problems in facing the future. In retrospect, it is evident that only our reluctant success to the Soviet thrusts posed by the possibilities of first a "booster pop" and then a "main site pop" was responsible for development of the superior rocket strength we have today. Too many of our leaders are only lip-service to the idea of leadership and are reluctant to probe the future with much vigor until it becomes practically necessary for survival.

Because of our superior technical, industrial and managerial resources, we have so far managed to come from behind in most of the critical technical areas of the post-war era.

But in an age when the possibilities of technical surprise lurk behind every laboratory door it is becoming an increasingly dangerous method of charting the course of a great nation.

—Robert Slats



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Washington Roundup

Kennedy-Vinson Clash

House Armed Services Committee is on another collision course with President Kennedy because of disorientation with the way Defense Secretary Robert S. McNamara is running his department.

Unlike the recent B-70 fight, which was settled amicably when Committee Chairman Carl Vinson and President Kennedy met at the White House race track, this battle is a close matter of principle with no confining technical considerations.

Rep. Vinson is pushing legislation through the House which would set the terms of all members of the Joint Chiefs of Staff at four years instead of the customary two for the Air Force, Navy and Army chiefs. The Marine Corps commandant already has a five-year term. The President still could remove the chiefs at any time under the bill.

Backers of the bill contend the four-year term would make the chiefs less inhibited when asked by Congress to express their own views on military issues. The legislation contained the inkblot of President Kennedy to reappoint Adm. George W. Anderson to another five-year term as chief of Naval operations and the reappointment of Gen. Curtis E. LeMay to only a one-year term as Air Force chief of staff amounted to punishing them for opposing McNamara.

Members of the House Armed Services Subcommittee, during hearings on the bill last week, bluntly told Deputy Defense Secretary Russell L. Gilpatric that the reason was a "reaction" to the dictatorial way McNamara was handling the joint chiefs. Rep. Levitt Aronoff, ranking Republican on the full committee, said "It is not so much what the joint chiefs tell us as what they don't tell us." He said he was supporting the bill to create "an independent military body."

Gilpatric and the Administration opposed the measure because it would reduce the President's flexibility in dealing with the joint chiefs. He said presidential removal of a chief before his last year term expired would carry a stigma for being discharging this inkblot of responsibility. He denied that the chiefs are inhibited under the present action.

Rep. Vinson and his colleagues on the armed services committee will continue to push the bill through Congress despite the threat of a presidential veto. The bill is providing a forum for Congress and military officers to challenge McNamara's stewardship of the Defense Dept.

B-70 Outlook Dismal

Defense Dept. officials are now expressing doubt whether the USAF-North American XB-70A Mach 3 aircraft will ever fly, but they do not explain whether they base this conclusion on unassessable technical difficulties or financial shrewdness. As of last week, wings had not yet been ordered to the final stage of the first aircraft because of fuel tank leaks and wing involvement. Air Force announced in August the refund would be late this year. Shippage of the refueler and first flight dates into next year and slow financial shrewdness with interest last spring by Air-Systems Division & Boeing Transportation (May 15, p. 25). Original first flight date was December, 1962.

Now it appears that the refund will slip well into next year with a first flight next spring. Aircraft number two and three are supposed to follow at nine-month intervals.

Slayton's New Role

Donald K. Slayton, who resigned his commission as an Air Force major effective Nov. 28 to supervise astronaut activities in a space agency civilian (AW Feb. 28, p. 18), was chosen by the selection board that joined the third group of astronauts, the last until 1967.

The new group of 14 astronauts includes a PhD and another who is working toward it. The group is younger and better educated than the 16 astronauts now on the rolls, and has considerable flying experience. Breakdown of affiliations of the new group: Air Force, seven; Navy, four; Marine, one, and civilian, two.

Wish for the final Project Gemini report to emphasize the need for increased astronomical and propulsion research to support a larger USAF role in providing tactical support and large-scale uplift, as well as another generation of manned strategic aircraft for the 1970s. USAF project leaders fear Defense Secretary McNamara's holistic leaning on current nonaerospace last work in Los Angeles.

—Washington Staff

NASA, DOD Agree on Space Station

Washington—Defense Dept. and National Aeronautics and Space Administration last week agreed to coordinate their growing space station study efforts, and at the same time laid the groundwork for the President to direct if a space station program should be undertaken and who will manage it.

The agreement, signed by Defense Secretary Robert S. McNamara and James E. Webb, NASA administrator, provides for coordination of all space station studies through the existing Aeronautics and Astronautics Coordinating Board.

The AABC is to make recommendations to Webb and McNamara, who will advise President Kennedy on the need for a space station program and its management. NASA-Defense Dept. efforts can be applied to the President.

Both space station programs in progress and management responsibilities, as discussed and assigned a joint NASA-Defense Dept. group, similar to the Project Gemini Planning Board, will be formed to establish objectives and select equipment.

NASA is spending more than \$2 billion on 36 space station studies (AWF July 27, p. 58) and the Air Force has reserved both the new parallel study contract to be awarded early next year (p. 26) on the combined military space station (CNSA).

NASA-Defense Dept. agreement means it demands from Congress for a military space station program as necessary. With the research how landing program requiring a NASA budget of between \$5.56 billion for the next few years, joint NASA efforts are doubtful that the agency should undertake a space station program concurrently with April 1968.

Air Force, on the other hand, considers the space station as its top priority space program and is arguing that the nation needs an operational station by 1967. Agreement reached last week may resolve what says NASA will take care of its manned orbital research laboratory (MORL) and the Air Force on its NSAS program (AWF Oct. 14, p. 25).

USAF Launches Two Vela Hotel Satellites

Cape Canaveral—Air Force launched a pair of Vela Hotel deep-space nuclear detection satellites here last week after troubles with the Air National Guard's second stage delayed earlier attempts.

Also included in the package was an advanced tetrahedral research satellite, the fourth in a series of small payload satellites designed to make radiation measurements.

The two 487-lb satellites, carrying 12 X-ray detectors on the 12 points of their acorn-shaped shape, and gamma-ray and cosmic-ray sensitive silicon shell, were launched at 9:12 a.m. EST Oct. 16 after four postponements.

The tandem-mounted satellites were launched on an expedition from the Agency in 10 minutes from 57,000 ft., the apogee of a high, elliptical orbit. One of the two satellites flew into a small solid-propellant motor contained within the satellite and into a circular orbit at an altitude. The other satellite was to have made one complete pass in the 280,570,000 ft. elliptical orbit and, as it came back to the target point again, it was to have fired its reaction motor. As it two now pass a circular 57,000-ft. orbit, the other would have moved 100 deg. away.

Large phase difference between the satellites was intended to preclude their making the same station and other cosmic phenomena as nuclear explosions.

B-58 Record Flight

Washington—U.S. Air Force set the new speed record between Tulsa and London as the result of an Air Force General Dynamics B-58 flight on Oct. 15 as it crossed the East coast of the United States. The B-58 was flown in 8 hr. 35 min. at an average speed of 956 mph.

The breakers, commanded by Maj. Robert G. Kibicki, showed to achieve record speed in the East coast of the United States. The B-58 was flown in 8 hr. 35 min. at an average speed of 956 mph.

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B-52 Follow-on Funds May Be Sought in '66

San Diego—Fast heavy funding for development of a follow-on aircraft to replace the Boeing B-52 might be requested as fiscal 1966. Under Secretary of the Air Force Headquarters McMillan said at a press conference here last week.

McMillan indicated that he expected a follow-on aircraft to be in operation before the final phase-out of the B-52 fleet in 1972. B-52 will be in operation for that time period with several stations reducing the size of the fleet, he said. The two types of follow-on aircraft were listed by McMillan as:

- Large, multi-purpose, long-range aircraft capable of such missions as air defense, airborne control center and remote launching platform. Such an aircraft is currently referred to as Magic, a new under study by USAF (AWF Sept. 16, p. 26).

- Low-altitude manned penetrator. Air Force is studying such an aircraft formerly called LAMP and now called AMP, an advanced unmanned penetrator (AWF Oct. 7, p. 21).

McMillan was a keynote speaker at a symposium on ballistic missile and space technology sponsored by the USAF's Space Systems Div. and Defense Systems Div.

Ling-Temco-Vought Changes Structure

Dallas, Tex.—Ling-Temco-Vought has reorganized its corporate structure, dividing the company into three divisions: Ling-Temco-Vought Corp. and Temco Electronics & Machine Co., retaining 10 divisions and has given a number of divisions new posts and titles.

The change is not expected to affect the company's traditional areas of work, but it is aimed at strengthening intercompany lines of communication and reducing costs by eliminating some overlapping administrative duplication, according to James J. Ling, who has assumed the post of chairman of the board while retaining chief executive.

Robert McCulloch, former board chairman, will retain his post as a chairman of the LTV electronic companies. McCulloch will be stepped down as board chairman to relinquish day-to-day responsibilities, but he will continue to act as an active and participating member in the company's general business affairs.

Paul Thrift, former president of the Ling-Temco-Vought Co., was made senior vice president of LTV. He will direct LTV's operating divisions: LTV Aerospace, LTV Electronics, LTV Radio Systems, LTV Michigan and Kinross-Harris.

Ling-Temco-Vought's electronic and



Fairchild May Build Turbo-Porter in U.S.

Fairchild Aircraft has signed a license agreement with Fokker Aircraft Works, Stockholm, Sweden, providing for the sale of the Swiss company's Turbo-Porter (AWF Dec. 7, p. 92) in the United States. Fairchild currently has one aircraft for demonstration purposes and two others in stock. West Africa and Northern Continental have had the aircraft in service since late 1962. Swiss-built version is powered by a Turbo-Porter engine, the Turbo-Porter M 400, has a maximum speed of 90 ft. 101 in., a height of 10 ft. 6 in. and a gross weight of 4,320 lb. The aircraft currently is being demonstrated to the U.S. Army.

consular divisions will report to Executive Vice President Clyde Stone, who has been named chief financial officer. Stone and Thorpe report directly to LTV President and Chief Operating Officer Gilbert K. Johnson.

All LTV operations were grouped into 13 divisions. Each of the 13 divisions heads has been promoted to a corporate vice president in addition to being general managers of the newly named divisions.

These are: Fred Reberling, LTV Texas Instruments Div.; J. Russell Clark, LTV Aerospace Div.; D. G. Gibson, LTV Vought Aerospace Div.; W. R. Kiefer, LTV Michigan Div.; and J. D. Weldon, LTV Central Electronics Div.

Other newly named divisions are LTV Air, consisting of Alvin Lanning and in Fokker Aircraft Products Div.; Alvin Lanning, LTV Aerospace, LTV Air, Alvin Lanning, LTV Aerospace, LTV Radio Systems, LTV Michigan and Kinross-Harris.

C-W Extends Garrett Stock Plan

Curtis Wright Corp., after failing to acquire all 700,000 shares of common stock, sought in the Curtis Wright Corp. was extended to offer and extend the new price from \$50 to \$57 per share.

Curtis Wright said the offer would be withdrawn Oct. 11 and would not be renewed. Shares offered to Curtis Wright since the offer was extended (AWF Sept. 16, p. 99) will not be accepted, but owners may take advantage of the higher price, the firm said.

Although a "very sizable number" of shares had been tendered under the original offer, Curtis Wright acknowledged that the amount "was not large enough to warrant paying the premium price of \$58 a share." The total offer of 700,000 shares represented about 47% of outstanding common stock.

Garrett has opposed the C-W plan and has sought a federal court order to prevent any stock acquisition.

Curtis Wright's first offer expired Sept. 27. It was extended to Oct. 11, after which the company had the working plan in which to decide a course of action. Garretts will review its latest position on the five-day offer Oct. 11.

As a part of its new offer, Curtis Wright said Garrett shareholders are invited to sell in 10 days prior to the offer on acceptance of the offer. Curtis Wright said the premium was an offer because Garrett, after last offering of the plan, had moved as usual meeting from Nov. 27 to Nov. 6 "on a temporary period."

Garrett had a strong offer in its hands with Curtis Wright. One of the prior to termination of the first offer. Garrett Oct. 11, the firm said, had purchased over 100,000 shares of Garrett stock for investment purposes.

Segal, a producer and controller of common stock, said that the offer was extended to Oct. 11. He said that the offer was extended to Oct. 11. He said that the offer was extended to Oct. 11.

A Garrett spokesman said that Segal's purchase removed those shares from possible sale to C-W. In addition, the announcement that Segal was buying the shares had an important legal aspect on the final day of the offer, he claimed. It is believed that some holders of large blocks of Garrett stock wanted until the last day before deciding whether or not to submit shares to C-W.

Prior to the C-W tender, Garrett stock had sold at around \$42. During the period of the offer the price on the New York Stock Exchange was between \$48 and \$50.

Segal's recent purchases probably will be the owner of the second largest block of Garrett shares, according to Garrett. Largest single holder is Merrill Lynch, Pierce, Fenner & Smith, which prior to the C-W offer held for staff and an investment about 17% of the 1,494,000 Garrett shares outstanding.

Space Weapons Ban

United Nations, N.Y.—U.N. Disarmament Committee last week asked that a resolution to the General Assembly calling for a ban on nuclear and all other types of non-nuclear weapons in outer space.

The resolution, also aimed at prohibiting satellites of such weapons on orbit and below, was sponsored by 17 of the 16 nations—excluding the U.S. and the Soviet Union—associated on the committee. France, the only committee member which boycotted the General Assembly meetings, refused to open the resolution but it is expected to vote for it in the General Assembly.

KC-135Fs to Support Mirage 4s

France-French air force will take delivery in January on the first of 12 Boeing KC-135F tankers which will be used to improve the on-air refueling capability of its Mirage 4 strategic bombers.

It is expected that the French will get the additional tanker at the rate of one per month, with all 12 received in their hands by 1984. A French tanker crew can spend training on the KC-135F test aircraft in the U.S. Eight additional French air bases capable will have completed similar training in the U.S. by the end of the year.

France's Strategic Air Force, being put together around the Mirage 4 bombers, will upgrade the tankers. Its transport division also will have six of the tankers in their present version, in use as an urgent military transport need.

The French air force also is taking delivery, on schedule, of its first Mirage 4 bombers. Scheduled crew training has been under way on two Mirage 4 prototype tankers. It is expected that the first Mirage 4 squadron will be fully operational by mid-1984.

A platformer both aircraft for the French strike force has been ready for some time. Studies started in 1977 are being studied for the Mirage 5 and Stratosil fighters.

Patent Rules Designed to Eliminate Variations in Government Policies

Washington—Aspatent protection no longer will find such wide variations in the policy of National Aeronautics and Space Administration and the Defense Dept. if the current policy guidelines issued by President Kennedy produce the intended effect.

In issuing the guidelines Oct. 10, President Kennedy said, "It is not feasible to have complete uniformity of policies" in deciding whether the contractor or the government should take title to inventions made under federal contracts, but "there is need for greater consistency in agency policies."

The guidelines attempt to protect the interest of the government and public without denying contractors the choice to exploit commercially those inventions they made largely because of their own special competence in a given field. The impact of the broad objective will not be felt until the federal agencies actually make patent contracts with current Defense Dept. agencies, which currently let the contractors keep title to inventions. NASA normally does the opposite by letting title to its contractors' inventions. NASA officials have complained that this policy did increase makes it easier for DOD to attract contractors for research projects.

The policy guidelines state that the government "shall usually acquire" title to principal rights in inventions made under federal contracts where:

- "A principal purpose of the contract is to create, develop or improve a product, process or method which is considered by commercial or other persons as having economic value."

- "A principal purpose of the contract is for exploratory work which is likely to generate the public health or public safety."

- "The contract is in a field of science or technology in which there has been little significant experience outside of work funded by the government, or where the government has been the principal developer of the field, and the acquisition of exclusive rights at the time of contracting might result in the contractor's performing as dominant position."

The acquisition of the contractors' title to the invention of a government-owned research or production facility or for conducting and deriving the work of agencies.

However, the head of an agency or department could give the contractor special rights "as necessary to enable him to carry out his contract and to give him the right to exploit commercially the invention he made largely because of his own special competence in a given field. The impact of the broad objective will not be felt until the federal agencies actually make patent contracts with current Defense Dept. agencies, which currently let the contractors keep title to inventions. NASA normally does the opposite by letting title to its contractors' inventions. NASA officials have complained that this policy did increase makes it easier for DOD to attract contractors for research projects.

The new policy guidelines state that the contractor "shall normally acquire" title to principal rights in inventions made under federal contracts where:

- "A principal purpose of the contract is to create, develop or improve a product, process or method for use by the government, and the work of the contractor is in a field of technology in which the contractor has no

significant technical competence (determined by factors such as known, experience and patent activity) directly related to an area in which the contractor has an established own governmental commercial position."

The government in the field and other cases would not insist on at least one commercially successful, commercially successful.

Even if the technical competence of the contractor did not fit the above description, the agency could still grant him title to the invention or other special rights if it determined this would best serve the public interest.

Any contractor contract, whether at special rights to an invention must develop it "in the past of practical application" within three years or face the possibility of the government becoming someone to market it. The government also could license more than one contractor of the public interest to develop a major line of production.

In cases where the government took the principal U.S. rights to an invention but not the foreign ones, the contractor could be given foreign rights as long as this did not conflict with treaties.

The guidelines say "Government-owned patents shall be made available and the technological advances covered thereby brought into being in the shortest time possible through dedication or licensing and shall be listed in official government publications or otherwise."

New Mooney Mark 22

Kennett, Mo. is to deliver a four-seat, 210-hp, single-engine aircraft to the U.S. Navy in 1983. The aircraft was developed in 1963. It was studied here at Mooney Aircraft, Inc.'s 1984 studies and distribution meeting last week.

Company also indicated plans for a more powerful, light version of its present Mark 22 four-place aircraft.

Mark 22 has been built for many years, and the Mark 22 prototype aircraft, and several versions of the design have been built. A prototype is scheduled to begin flight tests in October 1984. The aircraft will be powered by a new 210-hp, single-engine, four-cylinder engine. The Mark 22 is planned to be built in 1983.

Mooney's Super Mark 22 is a 200-hp, Lycoming four-cylinder engine, which increases horsepower by 10 hp over the present Mark 22 and adds about 12 mph speed. Price is estimated \$24,000. Mark 22 will be built in 1983, compared with \$14,000 in the present Mark 22.

Mooney also produces the four-place Mark 220 which adds 141 hp, 991 and will assemble the Mark 220. The company is in a field of technology in which the contractor has no



CX-4 HEAVY LOGISTICS transport design developed by Douglas is shown in artist's concept.

CX-4 Concept Has Swing-Nose, Aft Ramp

By C. M. Platner

Los Angeles—Douglas Aircraft Co.'s concept of the proposed CX-4 heavy logistics transport calls for a two-engine airplane with a swing nose and aft ramp, permitting simultaneous loading or unloading at both points.

Once viewed primarily as a replacement aircraft for the Douglas C-119, the CX-4 design's basic mission now, it is matched as a potential light transport at the speed of global deployment of U.S. based Army troops and equipment. The design also has a capability for utilizing complete Army divisions, in closing out operations.

Growing pressure from congressional and military sources is being directed toward development of increased airlift capability (AWP Sept. 30 p. 25) in the program.

Indications of Defense Dept. interest in the CX-4 came last week when representatives of the Air Force Research Laboratory and in San Diego first DOD main body with focus on the CX-4 concept.

A House staff source subsequently has recommended that funds be made available to the Air Force for Special Operations Requirement (SOR) for the CX-4. The recommendation was made in a report of a special subcommittee on military aviation, and was recommended that \$30 million be requested by the Air Force for a Program Change, Program (PCP) be required by Dept. of Defense at the earliest possible date.

With money available for further study and implementation of a SOR, the work would be paid for possible Request For Proposals (RFP) which is expected early next year, according to sources' opinions.

Douglas has noted the CX-4 for a 4,000-sq.-ft. mission with a 175,000 lb. payload. Used in conjunction with the Lockheed C-141, a combination fleet could deliver a fully-equipped Army division in seven hours by design range within five days.

Other areas of potential development over longer range legs up to 3,500 sq. ft. such as southeast Asia, also could be useful, although it reduced payload capacity. With a global airlift capability, U.S. troops could be withdrawn from many foreign bases and relocated in the U.S.

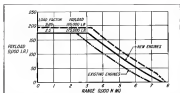
Douglas feels that if a development program was begun in the middle of Calendar Year 1985, the first aircraft would be built and flown in 1986. Aircraft could be in operation as early as the end of Calendar Year 1986, according to the timetable.

The Douglas presentation outlined the company's effort in studying the CX-4 over the past three years. The effort included fabrication of a detailed full scale wooden transport model, which has been valuable in studying loading and unloading techniques. The model version, built with a 36.0 ft. x 100 ft. fuselage, is known as the D-302 version of the CX-4.

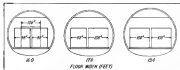
Current version of the CX-4 airplane is 17.5 ft. in 110 ft. fuselage and is known as D-305. Douglas details of the D-305 include:

- **Swing nose.** The nose section, which can be swung completely away from the fuselage. Control cables are carried across the hinge and pre-tensioned and are controlled by a single cable connected to the hinge point. The nose would be swung to the right side of the aircraft and could be opened in winds up to 40 to 60 ft. Advantages of the swing nose are the large loading distance and direct-through loading and unloading. As the loading door adds an approximate weight penalty of only 3.0% of gross weight. Doors of the emergency section are simplified from an aerodynamic drag and weight standpoint which would be considerations with a swing tail or lower aft ramp.

- **Fuselage width of 17.5 ft.** Although the wing span is built with a fuselage width of 17.5 ft. fuselage width is determined by a standard system of transporting palletized cargo. Fuselage width of 17.5 ft. is also acceptable in transporting lighter Army vehicles and military divi-



RANGE AND PAYLOAD performance of proposed CN-4 transport (as shown in chart above) with existing and new engine designs. Characteristics of three engine widths (below) have been considered.



seems. For heavier mechanical and structural dimensions, widths between 17 and 24 ft. prove feasible. The 17.5-ft floor width was chosen from three options—16.0, 17.5 and 19.4 ft—dictated by efficient loading of standard 45TL pallet loads (28 ft. 105 in.). Douglas feels a floor length of 110 ft. is optimum for a 17.5-ft width. This provides an internal volume of 26,705 cu ft, including the aft ramp area and 24,800 cu ft usable cargo in the three lower decks.

• **Design range** of 4,000 mi. in. This would provide direct aerial capabilities from U. S. to Europe. However, to achieve global aerial capabilities, including Southeast Asia, displacement ranges from 5,000 to 5,500 mi. are necessary. Douglas' payload versus range studies indicate that a CN-4 designed at a 4,000-mile-on-cargo aircraft could deliver a 75,000-lb payload (2.5 load factor) or at 50,000-lb thrust engines on a 5,500-mile-on-cargo. The same aircraft could deliver a 135,000-lb payload 4,800 mi. or a 175,000-lb payload approximately 5,100 mi. on.

• **Capability** to operate from support airfields to provide an efficient en route capability as a global base. The CN-4 will have to operate into and out of unimproved fields of 3,500 ft. or more. One is difficult at defining a standard runway condition for such airfields defined by the Army Corps of Engineers as support airfields, Douglas has not defined a landing gear in detail

padding further amplification of requirements. But it is confident that the design of a jet to meet the various and equivalent is feasible. In preliminary work on landing gear configurations, Douglas has used various conditions ranging from CBR (California bearing index) of 12 to 20.

One representative design for a CBR 16 or better surface condition is a gear carrying 16 17.5-in. main tires with two sets of dual tires for each of four struts. Tires would be inflated to 125 to 135 psi for takeoff and inflated to 80 to 85 psi for landing. This would require an on-flight airfield system, which Douglas feels is feasible.

Destination Performance

Destination performance has been used for loading a 160,000-lb payload in 5,070 ft. plus sufficient fuel for a 3,400-mile-on-cargo trip. Takeoff distance after offloading the 160,000-lb payload is calculated at 2,750 ft., or including additional fuel for a 4,000-mile flight. All fuel tanks would be located in the wings.

The Douglas CN-4 is designed to cruise in the Mach 0.88 speed range at approximately 10,000 ft. Takeoff weights referenced by Douglas for its 16.0-ft, 17.5-ft and 19.4-ft floor width versions are 157,600, 165,000 and 218,000 lb respectively. Calculated takeoff distance with a design payload weight of 135,000 lb, 7,300 ft. on a 90° dis-

The 17.5-ft floor width version has a wing span equal to its length of approximately 280 ft.

The forward ramp has an 11-deg rise and the aft ramp angle is 13 deg due to a 1-deg rise in the floor from nose to tail. The aft end, with a 9 x 10-ft. ramp opening, actually has a vertical clearance of over 12 ft. when the ramp is lowered from the horizontal position to the ground for loading.

Size and number of powerplants used on the CN-4 will be determined by requirements specified out in the yet unwritten specifications. Douglas can use propulsors between a new 30,000-lb thrust engine engine, although the company has studied one of existing engines such as the Pratt & Whitney TP-31, which would develop 21,000 hp of thrust. Use of lower thrust engines reduces the design payload capability from 135,000 lb to approximately 105,000 lb (and 4,800 mi. to 3,400 mi.). Douglas is currently considering using an engine on all three possible floor width versions.

System Cost

As viewed by Douglas, development of the CN-4 for the troop and equipment deployment mission is not only technically feasible, but not only cost-wise, but also cost-effective. A basic problem in developing an Aero derivative is the relatively low density of the cargo. With existing aircraft, low floor loads from 35 to 12 ft., the average floor loading for Aero deployment missions is 40 to 90 psf. The comparison with 110 to 150 psf for cargo transport.

Thus the CN-4 would be a primary carrier of low-density cargo, which it has been used for, leaving the high-density payloads for the C-141.

A fleet of CN-4 and C-141 aircraft consisting of approximately 125 of each would be an optimum fleet according to Douglas studies. Although operational requirements are as defined by the present need for this air, although it is based only on best estimates. This studies also show that approximately one-third of a required Aero division in terms of weight would be carried by a C-141. Composition of the division for study purposes was composed of 1 infantry, 1 command and 1 armored battalion.

A proposed troop/cargo version of the CN-4 would be able to carry 126 troops in an upper deck, extending forward from the midpoint of the aircraft. Troops would be seated backward, second deck. Infantry could be seated in the main cargo compartment, which has a 1-ft. ceiling-floor clearance, below the troop deck. Troop entry can also be managed by consulting the two full-length airlocks on either side of the fuselage into both ends.



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THE BOY IN THE DARK TRUNKS

LEACH RESTATE OF THE AIR—25

"Air battles regularly took place over the state lines, where tens of thousands of soldiers of both sides took the air, as a kind of theater. Or perhaps more like an aerial boxing ring—the troops saying, 'That's my boy in the dark trunks on the left!'"

The man who wrote those words is Raymond Collishaw. Twenty years ago he was Flight Lieutenant Collishaw, Commander of the Black Flight of Number 30 Naval Squadron, themselves of 60 German planes, and a great British ace of World War I.

The Black Flight was made up of men, who, like their leader,

were Canadian in their early careers. The planes they flew were Sopwith Triplanes dubbed "Black Ropes," "Black Death," "Black Sheep," and "Black Prince." Collishaw picked the "Black Death."

Powered by a 130-h.p. Gipsy motor, engine, the Sopwith Triplane was one of the most maneuverable fighters of the time. It was 15 feet 30 inches long, had a span of 35½ feet on all three wings, and was armed with one fixed synchronized Vickers machine gun in front of the cockpit.

The most remarkable feature of the Sopwith Triplane was its

dividing ability. It could reach 15,000 feet in 19 seconds! Collishaw and his Black Flight were credited to 13,000 lost planes a day. They'd say up themselves in argument-working for German planes to shoot down.

And about them down they did. By June 16, 1917, the five young Canadians had shot down a total of 36 German planes. In June and July of that year, Collishaw had 25 kills to his credit.

Unlike Rollo-Rose's "Flying Circus," whose tactics were largely defensive, the Black Flight sought out the fastest German planes it could find. Fluffy and snappy, Collishaw's Canadians were looking for a fight.

And they fought well. So well, three didn't have one casualty until June 26, 1917. On that day, they met up with Lieutenant J. E. Nash in "Black Sheep" was reported into the group and attacked by one Allister, one carrying Robertson himself, other Lieutenant Neil Allister, another two were in the German squadron. Allister really outmaneuvered Nash, and after a long battle from his superior gun, and the Canadian evading in the ground. Nash lived, though, to spend the rest of the war in a German prison camp.

The Black Flight was best on its revenge. And the next day, they went hunting from this flying group, 19,000 feet up, they spotted seven Allister's planes below them. They were part of Lieutenant's Jagdflieger 11, crashed German well with flying the last plane was Lieutenant Allister.

A pre-arranged plan, Collishaw gave the Black Flight the signal to attack. He flew in the middle with two Sopwith on each flank.

Because of previous losses with Jagdflieger 11, the Canadians knew the danger of flying below it. So, they dove straight in the German formation, began firing at long range, and kept up the barrage until, at the last possible second, they pulled up just above their targets. Their tremendous speed carried them above the Allister's again, ready for another attack.

On that first dive, Collishaw headed for the lead plane. After he fired for a few moments, Allister dodged to his side, began to spin, and whirled to his death. Nash was pinned.

Collishaw kept heading off German until September 20, 1918, when he was sent to England to help organize the Royal Canadian Air Force. He stayed until the end of the war.

At this time, the rival soldiers of the war were in a hot ditch struggle against the Bolsheviks. They needed air support. The British sent them a request recommended by Lieutenant Collishaw. When the Bolshevik regime

succeeded, Collishaw took off for Paris where many Reds were causing trouble. Finally, after six years of continuous fighting, Collishaw put his plane away and began living through the trials of Canada's postwar air force.

He put on the gloves one more time. As an Air Vice Marshal, he commanded the R.C.A.F. in the Middle East during World War II. Now living in Vancouver, British Columbia, the boy in the dark trunks has retired unadorned.

What's he doing now?

As the ace of 11, he's very active in the exploration and development of copper mines. He's Max Leach. He earned his reputation doing one thing: now, he's paid at another.

How did Leach earn its reputation?

By making precision claims for his aviation industry. He began back in 1939. We're still making claims, of course. But, like Vice Air Marshal Collishaw, we're very active in other fields, too.

What other fields?

Technology, for one. Leach developed a new way to measure telemetry transmission (shown) with the widest FM response in the industry. And for its advanced in tape recording capability, Leach was awarded the contract to develop laser light tape recorder for the three-way Apollo program.



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News Digest

Free American World Airways
Trans World Airlines finally announced this merger agreement last week. The agreement, pending since Dec. 10, 1962, had lost much of its appeal by last spring (AWM Mar. 4, p. 33).

International Business Machines Corp. will assemble, integrate and check out, installation and, guidance equipment for the Saturn IB and Saturn 5 rockets under a \$75 million, five-year NASA contract.

Two 6-wheeler vehicles, launched Sept. 15, 1962, stopped sending usable cloud cover pictures last week. NASA said that after 13 months of nearly perfect operation, a malfunction in the vehicle's camera from camera regulator caused it to send faulty pictures. Attempts were being made to diagnose and correct the trouble.

Dr. A. A. Goldfarb, inventor and developer of the Flying Bedouin virtual satellite and flying research vehicle, died Oct. 11 at 78.

AC Spark Plug, Raytheon Co. and the Rockwell International Corp. will develop and build navigational and guidance systems for the Apollo Lunar Excursion Module. The three firms under direction of the Massachusetts Institute of Technology are developing the Apollo command module guidance and navigation systems (AWM May 14, p. 27; June 24, p. 34).

Chief Air Marshal Pavel Fedotkinovich Zhigayev, who headed the Red Air Force from 1949 to 1957, died recently in the Soviet Union at 68. Zhigayev was son of chief of Armistice, the Soviet-owned airline, during the period from 1937 to 1939.

Two-man cockpit assignment of the Douglas DC-9 has been accepted by Federal Aviation Agency pending flight demonstration.

Massachusetts Institute of Technology Research Laboratory of Electronics (RL-9) has been accepted by Federal Aviation Agency pending flight demonstration.

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U.S. Carriers Support SST Development

Four airlines indicate intentions to buy 29 supersonic transports; Boyd warns Senate unit of pitfalls.

By Robert H. Cook

Washington—Four major airlines moved to support the nation's supersonic transport program last week by placing tentative orders for 29 aircraft with the Federal Aviation Agency, thus offering the first competition for the British-French Concorde supersonic transport.

Trans World Airlines, Pan American World Airways, American Airlines and Flying Tiger Line are seeking early positions on a future U.S. supersonic transport production line, underlining the airline industry's deep concerns over the competitive problems that could arise from the development race between the U.S. program and the British-French Concorde.

Announcement of the orders coincided with the opening of hearings in the Senate aviation subcommittee on the supersonic transport program.

Chronology of the orders, the high-lighted the future competition, but the airlines TWA and Pan American, which will be first to operate a super-sonic, and the concern of American over the prospect of a supersonic competitor on the long-haul transoceanic routes. Orders were placed as follows:

- **TWA announced on Oct. 14** that it has agreed to purchase six U.S. supersonic transports, and give FAA an initial down payment of \$600,000. The order called for the "first and three after every second place" in a total of six. TWA said it would accept a speed of Mach 2.2 for the aircraft although it hoped Mach 2.4 might be attained.
- **Pan American, the same day,** advised FAA that it would purchase fifteen of the aircraft, involving a \$1.1 million down payment. Pan Am pointed out that it had not been considered the first delivery positions. When the Concorde enters its Concorde line first, Pan American contends it made the first offer to purchase the 48 supersonic jets prior to the TWA order. Pan American said, it repeated this offer to FAA but failed to receive an answer until the agency advised it had already given first preference to TWA.

The airline said it was willing to accept delivery of every second aircraft beginning with the first or second delivery, until its order for the transport was filled.

- **American announced two days later,** on Oct. 16, it had agreed to give FAA an initial \$1.5 million down payment toward the purchase of the first six U.S. supersonic transports, or as Oct. 19 telegram. At the same time the airline

said it had signed an agreement to purchase four Concorde's. The airline noted that should the U.S. abandon its own supersonic program, the Concorde order would be "good as cash." Grants noted that TWA, a company of American Airlines, and the West Coast, has ordered three Concorde from the British-French combine.

- **Flying Tiger notified FAA on Oct. 16** it wants to place a development order on two supersonic transports, but was confined over how soon to offer more than an "intent" from FAA.

Russia in SST Race

Washington—Russia is overhauling its own supersonic Mach 2.2 aircraft as the entry into world wide competition to produce the first commercial supersonic aircraft, according to General M. R. Degtyarev, administrator of the Federal Aviation Agency in charge of the supersonic transport program (AWF July 18 p. 31). Rost said members of a Soviet aviation subcommittee investigating FAA's supersonic transport program that the Soviet is too limited and was not so interested in post a challenge to the U.S. program as the British-French development of the Mach 2.2 Concorde. Part of the reason, he explained, is because of the Soviet's poor failure to sell their jetliner aircraft.

Rost told the subcommittee that FAA lacks the size of industry that a U.S. supersonic transport Civil and processing problems which have been high as the post war boom, however, and most have been about the size of industry now. Thus the aircraft industry has about the size of aircraft when the present subsonic aircraft were being designed, he said.

that order in the supersonic was being accepted. The subcommittee, which Flying Tiger claimed it had of the industry was indicated reports of the TWA and Pan American that it would require a convertible, convertible version of the supersonic.

Many supersonic orders may be announced in the near future, since FAA has contacted five of the domestic and international airlines but then, remains some uncertainty over whether or not the airlines will actually accept or accept down payments or attempt to generate problems on production lines.

As of late last week, FAA had not replied to the purchase offers, explaining that it must first ascertain whether it has the legal authority to accept the money. If it has the down payments will be placed in a special account with the U.S. Treasury Department.

Tatiana of CAB Chairman Alan S. Boyd indicated an airline fear that it from the risk of having an aircraft first that may be technically feasible, but operationally unsound. FAA has been pressing for a Mach 2.5 design speed and National Aeronautics and Space Administration for a Mach 3 speed.

General NASA, which proposed John Stuck, recently told the New York American that a representative from the Soviet Union is in contact with the U.S. government to discuss the possibility of that at today's national technology level, he designed for Mach 1.5 or higher. Stuck is now Republic Airlines vice president.

The object of producing a supersonic jet is not to overcome technical difficulties, but to produce a product which will serve the public most economically and conveniently. He said that the three companies, in the United States, which are producing a supersonic jet, are not yet ready to produce a product which will serve the public most economically and conveniently.

Although he stated the Soviet aviation committee of the Board's full support of the FAA program, Rost emphasized that "we'll take the leadership on supersonic" technology, he added that CAB is the best qualified agency to advise countries of the program.

Boyd said he considers the supersonic aircraft a logical extension of the aircraft that many countries are producing. He said the U.S. and the British-French are "wearing nose-colored gloves" in producing delivery dates on the new aircraft. More important than early delivery dates, he noted, is the need to make profitable operation.



Short Brothers Turbo-Skyvan Makes First Flight

Short Brothers and Bristol Turbo-Skyvan, which recently made its first flight (AWF Oct. 7, p. 37) shows new engine nacelles designed for Turbo-Skyvan 2 turboprop engines, which replaced the Concorde piston engines used in the earlier Turbo-Skyvan. Prototype will complete flight test and certification program with the 190-hp. Airframe 2b, but production aircraft will be powered with Airframe 30 engines of 417 hp. only. With the latter engine, the Turbo-Skyvan will have a cruise speed of 316 kt and will carry a maximum payload of 4,000 lb. Approach speed is 115 kt and landing distance is 1,800 ft.

Airline concern over delivery positions could cause a serious problem, he said. First, competition unable to secure favorable positions could result in "go home" switching decisions and be forced to compete with subsonic transports against supersonic.

No one carrier can be permitted to monopolize the delivery line and an equitable method of distribution is needed, he said.

Subsidy Payments

FAA does not intend that subsidy payments will be awarded by supersonic operations, but American Airlines that filed to purchase the Concorde might suffer large competitive losses that they would request subsidies for aircraft he indicated.

Boyd also feared criticism that the U.S. supersonic will be profitable over stage lengths of 1,700 mi. is "debatable," suggesting that profits will depend on longer routes. As an example, he cited the 1,100-mi. distance between Miami and Washington, pointing out that profitable operation would require Alaska-Washington-London routing.

In many cases it will not be possible to operate intermittent service on such short segments. He said extensive use of the supersonic transport will require a replacement of many existing major route aircraft.

This point conflicts with statements of General M. Rost, chief of FAA's supersonic transport program, who said a profit can be made on 1,000-mi. stage lengths (AWF Oct. 14, p. 34).

Supersonic transport firms must carry a exchange, for "once the ship is over" the public may not be willing to

pay the difference, Boyd said, adding that improved service has already demonstrated that new equipment to attract passengers.

To be profitable should be able to support the addition of the new aircraft, Boyd said since the CAB expects it will reach 77,000 or more twice the total of last year.

Boyd said CAB anticipates some foreign countries will integrate supersonic into U.S. supersonic operations in a competitive manner, but he is confident this could be solved. He said airlines must be needed to lead money for foreign countries to buy the U.S. built transport. He added that groups of time and these airlines can build together for supersonic operations in Europe and the Middle East and this arrangement could cause competition for U.S. flag carriers.

Operating Objectives

Proposed operating objectives of the U.S. aircraft also came in the strong criticism from Boyd, who pointed out the industry's lack of range and capacity as compared to the present subsonic transports.

The FAA proposal calls for a range of 4,000 mi. compared with the subsonic transports, which fly daily nonstop 4,500 mi. between New York and Rome. It would be a "step backward" if these transport services could not be maintained with the supersonic aircraft, he said.

"It would be unfortunate... to create a pattern of supersonic transports which caused a condition of busy-up and wait because of need for frequent

and lengthy fuel stops," Boyd said. Design period of the supersonic transport is also about 15% less than that of the present subsonic aircraft and should be increased to at least equal that of the present jets. In the past, Boyd contended, the airline industry has shown a tendency to achieve increased speed with an increased payload.

These problems have a direct effect on the industry's efforts to lower its unit costs, Boyd pointed out, and "carriers will be cautious in rebilling their increased in a single aircraft merely to make the travel time from Washington to London to London. Many less than one that particularly when it appears unit costs will not be reduced below those of jets."

N.Y. Helicopter Crash Investigation Speeded

New York—In addition to the crash of a New York Airways Boeing Vertol V-22 helicopter at Idlewild Airport, New York City, last week in hopes of getting the first look into airport as rapidly as possible.

The airline indicated suspended passenger service after the Oct. 14 crash which killed six. However, its remaining three Vertol helicopters were not grounded and were flying. A spokesman for the airline said they would be returned to service as soon as something definite was known about the cause of the crash.

wreckage of the helicopter was moved Thursday to La Guardia airport, where the airline has its headquarters

X-30 DYNA-SOAR. U.S. Air Force's X-30 spacecraft shown in diagram above as it will look in orbit before pilot begins controlled reentry into atmosphere for landing at airfield of his choice. Designed to explore problems of reentry from orbit and to develop technology of manned maneuverable reentry.

from space, Dyna Soar will combine speed of ballistic missile in space with controlled and accurate flight of an airplane in the atmosphere. From this and other studies, new space-oriented concepts may evolve. Among, an X-28 system contractor, will build the spacecraft and integrate the vehicle with its booster.

Capability has many faces at Boeing



NOBINO 727. America's first designer pillow, is powered along with Diving Vests 187 helicopter. Airlines have ordered 127 Boeing 727s. They enjoy service soon.



NATION & LOCATION Italy/Italy, U.S. Navy's amphibious assault ship, is shown "docking" an underwater "cannon." Length is 174 feet; speed over 40 knots. Role is: Escort, High Force in underwater operations used by U.S. Navy.



CREATING LABORATORY. Drigone's concept of orbiting space laboratory, based on Boeing studies. Boeing built NASA computer for research on orbital school laboratory and space vehicle in free mock and supplies in zero-gravity and orbiting space station.

At the same time, British Eagle, headed by Harold Bushberg, former chairman of Coastal Eagle, asked the court to restrict REA authority on Loe.

Nor Khan was accompanied by J. S. Maza, the artist's commercial director and regular delegate to the IATA annual general assembly.

At the same time British Eagle, headed by Harold Bursberg, former chairman of Caledonian, asked the board to reject RFA, exactly as Lord

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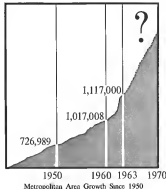
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BELFAST TURBOPROP TRANSPORT, built by Short Brothers and Harland, is scheduled to fly in November or December.

Two Aircraft Scheduled for Use In Belfast Flight Test Program

Bellon-Two aircraft will be used in flight testing the large Short Brothers and Harland Belfast turboprop transport (AW Sept. 9, p. 38), with the aim of achieving initial and civil certification in 35 to 38 months and 853 hrs. of flight time.

The plane, one of 10 ordered by Royal Air Force, was rolled out last month (AW Oct. 14, p. 37) to Short Kolls-Rover. Two engine run-up tests and two taxi flight tests will be in late November or early December, three years after the order was placed by British Ministry of Aviation.

Three more Bellons are in the final assembly line, with wings attached. A test flight was completed last January for stress testing in the wind tunnel at the company's Sudbury airfield.

In-flight certification for the test program is divided into five sections comprising a single system. There are:

- Digital recording system capable of simultaneous recording of up to 3,000 digital quantities
- Frequency modulation system to record all high frequency data
- Direct readout trace recording system
- Audio-visual presentation for in-flight monitoring of safety parameters
- Airborne takeoff and landing recording system

The Belfast has a clear hull measuring 12 ft. square by 61 ft. in length, with an overall height of 15 ft. 6 in. and overall length of 35 ft. Cargo volume is more than 11,000 cu ft. The plane can carry up to 140 passengers on a single deck. Its design provides for a two-deck installation if necessary.

The Royal Air Force aircraft is designated Short SC5/10. Maximum payload is 50,000 lb., and gross weight is

225,000 lb. Short Brothers also has designed an improved version for the transatlantic cargo market designated SC5/11, using four 131-hp engines. Maximum payload is 75,000 lb., and gross weight is 250,000 lb.

The third version is the jet Belfast transatlantic cargo freighter (AW June 10, p. 35). Jet version uses two engines. The same fuselage as the turboprop Belfast, plus the same wing used on the Lockheed C-141 Starlifter and a redesigned T-tail section. Engines would be Rolls-Royce Conway RB39-41 or 43 turbo-prop engines. Maximum payload would be 125,000 lb., gross weight 350,000 lb.

Approach Decision Cited in Slick Accident

Washington—Crew's decision to continue an approach after losing visual reference to the ground has been cited by the Civil Aeronautics Board in the probable cause of the fatal crash of a Shell Aircraft 1840T Constellation last Feb. 3 at San Francisco International Airport.

Two crewmen and two passengers were killed when the aircraft struck approach lights while attempting a landing. Three other passengers and one crewman survived the crash.

As a contributing factor to the accident, CAB listed inadequate monitoring of the instrument approach by the captain's previous approach radar observation.

Although the Shell aircraft had an insensitive glide slope receiver, it failed to carry an traffic control of the malfunction and elected to make an approach solely on the basis of radar ob-

servation from the previous approach radar controller, CAB said.

On final approach to runway 25R, the flight was advised that it was within high on the glide slope and to the right of the boundary crossing. The last PAR advisory placed the flight 100 ft. left of course passing the glide slope marker and 25 ft. above the glidepath. CAB said.


Radar control was terminated at that point, the Board said, despite the fact that the flight had not reported leaving the approach lights or runway in sight. Accepted procedure for an instrument landing system approach requires that the controller continue to monitor the aircraft position and advise the pilot whenever radar indicates an situation that might endanger the flight. This was not done and the radio services was terminated without advising the pilot, CAB said.

The aircraft struck the approach lights at a point 100 ft. below the glide slope and 1,000 ft. from the end of the runway. CAB concluded that had the controller have maintained the flight through the approach he would have observed the aircraft's descent below the glide slope and would have advised the crew.

American Profits Rise

American Airlines net profits for last nine months of 1965 at \$12,015,995 were double those for the same period last year, the carrier reported last week.

Included in last year's non-aerospace figures at \$6,817,339 was \$1,614,000 loss gain in sale of property. This year's figures included 159,000 of such profits. Non-aerospace revenues rose in 1965, 77,746 from last year's \$149,979,867. Operating expenses rose from \$292,626,895 to \$294,654,181, and total expenses from \$344,961,739 to \$392,491,185.



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SHORTLINES

► **Algeria and Ghana** have signed a bilateral air transport agreement calling for a direct air route between Algeria and Accra.

► **American Airlines'** freight revenue rose 10% for September over 2% over the same period last year, and 5% for the first nine months of 1963, compared with the same period last year.

► **Bonanza Air Lines** will increase scheduled mileage 26% during November and December through the inauguration of new flights.

► **Customs declaration forms** will be eliminated by U. S. Customs beginning Jan. 1. Thereafter, passengers entering the U. S. will make baggage declarations only at all airports of entry.

► **Continental Air Lines** has added the Civil Aeronautics Board for its rates and minimums at its three cities has plus. Gates and that on all Aug. 24 when the plan became one year old. Passenger revenue rates had climbed 25% during the 12-month period, compared with the same period in the previous year.

► **East American World Airways** will cut fares between Seattle and Alaska in half for weekend travel. The 74% over sea rates will apply between Nov. 1 and Mar. 31.

► **Southwest and East** passenger-loading schedule timetable presently planned for the Los Angeles International Airport has been placed in operation. Bidding will be placed by Continental Air Lines, Delta Air Lines, Pacific Air Lines and Pacific Southwest Airlines.

► **Texas-California Airlines**, an inter-state carrier, has been granted long sought approval to serve San Francisco International Airport and hopes to be the carrier only in November. Exact date of initial service awaits completion of insurance and bonding arrangements. Current presently operates three round trips daily with Lockheed Constellation 740s between southern and southern California, serving Oakland in the north and Los Angeles, Burbank, and San Diego in the south.

► **United Air Lines** and San Francisco Public Utilities Commission are negotiating a \$450,000 moving schedule at the airport. International Airport (AW) [see 24, p. 41] to accommodate passengers arriving between the airport's new lobby and United's gate positions on the terminal's north concourse.

AIRLINE OBSERVER

► U. S. domestic trunklines have reported an industry net profit of \$28.7 million for the first eight months of 1963, compared with a net loss of \$11.6 million in the same period last year. Industry net profit for the month of August was \$15.6 million on operating revenues of \$227.5 million and operating expenses of \$206.9 million.

► Domestic trunkline traffic during September rose 15% over the volume loaded in September, 1962, but declined sharply from the peak loaded in August. Load factor for the industry in September was 74.4%, a 6% increase over the load factor of the same month last year. Load factor in August was 68.1%, highest load reached since June, 1962, when 64.3% was recorded as the trunkline monthly average. Available seat miles in September reached 65.6 billion, an all-time high for the industry.

► Alaska ranks its 26% profit gains in Air Union interest because its traffic and route growth has exceeded projections made when capital was established in 1959. Last year, for example, the Union carrier reported a 16% passenger traffic increase. Alaska argues the Air Union plan is a true case, but both two possibilities might never be, it is necessary to purchase a separate transport fleet that might be feasible only in a joint project, as urged by the two U. S. flag carriers—Pan American and TWA.

► Soviet Union plans to build a "shorter" version of its Ilyushin Il-62 jet transport. It will have 60 berths. Conventional configuration of the aircraft, now undergoing flight tests, will carry 108-152 passengers.

► Boeing 737 jet group, fuel control and fuel heater system are "unaffected" installation, accounting only two fuel supply lines for all three units—one incoming and one outgoing—thus saving about 15-20 lines over conventional separate installations of these equipment.

► Federal Aviation Agency Administrator N. E. Tishley is calling the proposed local service aircraft the "S888ST" for "small, slow, safe and subsonic transport." He is expected to announce that the aircraft be developed as a government-funded design competition (AW Sept. 16, p. 49). In making possible configuration, he said the aircraft probably would be a turboprop, would seat close to 50 passengers rather than 38, would cruise at about 125 mph and would sell for under \$500,000.

► Hawker Siddeley 748 twin-engine transport is being selected to operate the 21st Air Force's new base in 1,600 sq. mi. Resultant increase in maximum range to 1,750 mi. was obtained by extending wingtip tanks outward toward wing tips. Use of new angle cut tank system developed by Probert Research Co., Burbank, Calif., and manufacturing under license by British Aircraft Ltd., allows entrance with no access in structure weight.

► Local service airlines reported a 12.9% increase in revenue passenger miles in September compared with the same month last year. Available seat miles for the month climbed 9.5% and load factor rose to 74.4%, a 1.2% increase over September of last year.

► British Overseas Airways Corp. planning action was threatened by its own employees in 51 Boeing 737 jets between London and Phoenix at 50 stands, but for winter night flights. Air Transport Licensing Board approved the late, advertising and public relations pulled out all stops—and then pulled them back in. Because no flights are operated at these times and none are planned.

► Russia's "VVO Teleshipper" organization has launched an advertising campaign to sell teleports to underdeveloped nations. Full-page ad in recent issue of the Soviet magazine Vostok (Vostok) (Foreign Trade) features a picture of the Komsomol airport at Chirchik, Republic of Georgia, and says that "VVO Teleshipper" studies technical assistance in designing and construction of airports and airport installations and delivers all the equipment, equipment and materials necessary for such construction." Another large, Soviet-built airport near Soma, capital of Yemen, was opened last month.



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that in more than 2,000 departures, its DC-8's had the lowest percentage of delays over a month's period for any transport in the airline's history.

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ALPA Challenges Finding by CAB That Crew Erred in DC-7 Accident

Washington—Air Line Pilots Assn. has disputed the Civil Aeronautics Board finding that faulty crew technique during an abnormal approach was the probable cause of the fatal crash of an Eastern Air Lines DC-7 north of New York last Nov. 30 at New York International Airport (ANW Dec. 18, 1962, p. 15, Cir. 14 p. 33).

Flight 512, commanded by Capt. E. J. Bechtel, ALPA safety spokesman, encountered fog near the threshold of runway 8R, attempted a go-around, but struck the ground on a slightly too-high altitude according to a CAB accident investigation report. A total of 24 passengers and two stewesses survived the impact and subsequent fire. ALPA immediately challenged the Board's findings as "misleading." The association claimed that the crew was unable to steer the crash because it had not been informed of the runway weather conditions.

Knots Count

Resolving events preceding the accident, CAB and the last radio contact with the flight was at 9:43 p.m. shortly after it had passed the outer marker and had been cleared to land. The crew had agreed to the tower clearing the approach light runway. Captain Bechtel's flight had landed 1 mile after this radio contact reported track log positions along the runway.

Flight 512 failed to report anything for the weather radar, although two tower controllers saw the aircraft's tail section become disruptive in fog about 1.5 miles out of the runway. At 9:44 p.m. the tower reported that flight 512 advised when it was clear of the runway, but received no response. One minute later tower controllers saw a bright orange flash near the precision approach outer building, located about 800 ft. to the left and 1,200 ft. beyond the threshold of runway 8R. An American Airlines flight on instructions of the tower, overflew the area and confirmed the crash.

The aircraft's wreckage was found about 2,500 ft. beyond the instrument landing system touchdown point on the left side of the runway. Weather conditions immediately following the accident were reported by the Weather Bureau as "visibility 1/2 mi. in fog, surface visibility 1/2 mi."

The Weather Bureau's surface visibility reading was not transmitted to the tower or the aircraft prior to the accident, CAB said.

Although the flight was attempting a landing with some estimates of 1 mi.

visibility, two indications of poor visibility should have been heard by the crew during the approach, CAB said. One of these involved an advisory that reported a visibility of only 18-60 ft. while clearing the runway and the other was a jet flight that had requested a takeoff delay because of weather conditions.

These transmissions had all taken place on the tower frequency, which Flight 512 had been instructed to monitor before landing, according to the Board.

Near-up Altitude

Examination of the aircraft wreckage indicated the aircraft was at a 540-ft.-due measure altitude with landing gear retracted, flaps set at 20 deg. takeoff position and all four engines developing power ranging from 1,800 to 2,035 hp at impact with 2,800 hp displayed on a normal landing approach. CAB investigators said.

About 9 deg. of aircraft rotation would have been necessary to execute the normal approach, CAB said, or more power should have been used.

"The Board concludes that additional aircraft rotation was not effected due to a lack of accurate instrument ascertainment and that additional power was either not requested, or delayed because of other factors," the report stated.

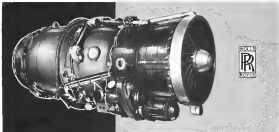
Inadequate Report

CAB emphasized that the fog had not been adequately reported to the flight, and pointed out that the Weather Bureau and Federal Aviation Agency had conducted an on-site investigation to correct the situation. Among these were recommendations that all terminal area weather conditions be reported to approaching aircraft, surface visibility range (SVR) measurements, respectively at the time of the accident be reported at New York International and that runway clearance be supplied when the SVR is out of instrument range.

ALPA contended that in its investigation disclosed that, "the probable cause of the accident was the flight crew's misreporting an unexpected and extremely dense ground fog condition that the aircraft had been erroneously placed into a short-landing condition. This loss of visibility necessitated a return to flight instrumentation reliance and the subsequent attempted go-around, in conjunction with the applicable manual, was responsible for the crash with the reference available," ALPA said.

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AERONAUTICAL ENGINEERING



PRELIMINARY ORDER for 134 of the Sub-105 fighter, shown above, was, in first flight photograph, has been placed by Swedish Air Force subject to acceptance bids later this year. Two-phase Dodge RDT all-weather fighter interceptor is in development.

Viggen to Be Core of Swedish Air Force

By Warren C. Wetmore

Stockholm-Swedish Air Force plans to replace all present first-line combat and reconnaissance aircraft with the Sub-105 Viggen by the mid-1970s, and increase its reliance on a variety of missile systems.

First aircraft scheduled for phase-out in favor of the Viggen is the Sub-A12 Lansen strike fighter, all of which will be retired by 1970. Longer service is scheduled for the Sub-Dragon all-weather fighter-interceptor which is in limited service, the Falcon-mounted F35, is considered an adequate mainstay for Sweden's air defense up through 1975.

Material Cutting

Material cutting was begun at Sah Aircraft Co.'s Lindagård plant for the first Viggen prototype, which is slated for rollout and flight testing in the late 1975-early 1980 period.

Previously, the Viggen will be somewhat larger than the Dragon and is designed to "sweep the whole flight envelope" of that airplane, East German, Soviet's doctrine of engineering, and recognizing the need without having too many of its expensive combat aircraft concentrated in a few facilities in the nuclear age, the air force sought a solution to the problem of adequate dissemination. Studies showed that a VTOL aircraft would have to

be half again as large as a STOL to produce the same ground attack capability. It was decided best for Swedish purposes to design the Viggen for STOL performance, so it could operate out of the small, widely-scattered missile refueling throughout the country. The Viggen will require approximately less runway than the Dragon, Boeing said. Touchdown speed lower than the Dragon's 135 mph will be facilitated by blown boundary layer control on the flap of the aircraft's unusual forewing. Air for this purpose will be supplied by compressor bleed.

Angle of incidence on landing will be somewhat greater than that of the Dragon, which touches down at 12 to 14 deg, but the Viggen is designed so this landing angle will not be adversely affected. The delta cockpit is situated well forward of the forewing and the nose is dropped for additional downward visibility.

Forewing has a clipped delta planform and will be located close to the aircraft's center of gravity. It is designed to give controlled vortex flow over the main wing, which in turn will yield a continuous flow of the coefficient of moment versus angle of attack curve and that improve the aircraft's stability. In addition, inevitable interference of shock patterns between the wings will occur in supersonic speeds.

Features in the double delta pattern is the Dragon, except that the

forewing pattern is changed. Forward of the hook, the wing has a lower sweep angle than that after the hook, reverse that of the Dragon.

Vortex for opposite the tail is designed to give symmetry to the aircraft's vertical surface.

The Viggen will be powered by a supercruise version of the Pratt & Whitney JTSD turbofan engine fitted with an afterburner. In powerplant, bearing the Swedish designation, J40A, will be built under license by Svenska Flygmotor AB of Trollhättan (AW June 8, 1962, p. 17). The U S design action is JTSD-12.

Military Versions

Der threat of the JTSD-12, which powers the Boeing 707 jet transport, is 14,000 lb. While no official figures are available on the military version, some Swedish sources have indicated that it will be used at about 24,000 lb. dry. Afterburning will boost the thrust to 28,000 lb. Battle weight of the engine is about 2,500 lb.

It was selected because of its low specific fuel consumption, a high constant factor on language, low-level shock and maneuverability margins. On the other hand, the high thrust augmentation with afterburning suits it for fighter-interceptor application, which requires high speed and rapid acceleration.

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engine will help in landing on short, ice- and snow-covered runways during the long, cold Swedish winter.

These internal armament of the Viggen will be up to four 30-mm. cannons. The A37's four lighter version will also carry external stores of bombs, mines and unguided rockets. Its RB-304 missile will be used against naval targets (AVR Mar 3, 1968, p. 209).

Another sophisticated missile already conceived is not yet been under development by Saab for about five years. Designed for use against both land and sea targets, it will feature automatic all-weather homing plus greater range and destruction power than the RB-104.

The missile is reported to have four highly-swinging wings and control surfaces near the aft end.

Fighter-Interceptor

An all-weather fighter-interceptor version of the Viggen will be armed with Hughes F48s in four missiles on their egress. Aircraft will be completely integrated into the STRIL 60 automatic air defense system (AVR Mar 3, 1968, p. 209).

A sophisticated solid-state digital computer on board the Viggen will ease the pilot's workload. The Saab-designed and-built device—about the size of a transistor radio—will perform navigation and position determination tasks for control and subsequent calculations, fuel data, navigation, information control of the aircraft and data exchange with the STRIL 60 ground-based computer.

Overriding speed of the computer is 100,000 addition per sec. and its rapid access memory has a capacity of more than 5,000 words. Input-output systems are designed to handle both digital and analog signals and have a substantial growth potential.

Computer Advantages

One of the great advantages of a digital computer over an analog computer such as the one now flying in the Duxford is one of reprogramming without physically altering the unit. Flexibility of the aircraft can be increased merely by changing its computer-based programs to introduce the latest situation, test results and tactics.

Modular construction is employed in the computer to facilitate maintenance, with functions that can be swung out for access to components.

Long-life planar silicon components are utilized in the micro-miniature circuit.

The Swedish Air Force is responsible for industrial and financial management of System 37. It has had some new procurement guidelines in order to best

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■ For questions, descriptive folder or separate literature brochure please write to: Hercules, Electronics and Chemical Production Department, Hercules Powder Company, P.O. Box 10, 8—P.O. Box 10, (Tolani) 82146—or 3400 Wilshire Blvd., Los Angeles, Cal. — phone DUN 7-8131.



utilize the available resources, both in industry and in the Air Force itself. As System 37 project coordinator, Saab will coordinate procurement under Air Force supervision, making sure that equipment meets specifications and is delivered at the proper time and intervals within the fixed economic boundaries. In contrast to the previous arrangement in Sweden, where nearly all material was developed to the Air Force's direct order, System 37 components will be developed and purchased by the Swedish Air Force.

PERT in Use

To facilitate chronological coordination of the work on System 37, Saab made a detailed, two-year study of the U.S.-developed Program Evaluation and Review Technique (PERT), and used it experimentally in connection with the Saab 105 jet fighter. Presently PERT is being used for planning and follow-up on System 37, and is expected to be a valuable tool in the program ahead.

Drakes version now ruling of the line in Europe are the F10D and S34E, the latter having a modified nose section for accommodating aerial reconnaissance cameras (AVR Jan 15, p. 30). These Mach 2 aircraft are powered by Rolls-Royce RB 144 Avon 500 engines built under license by Svenska Flygmotor.

Drakes are currently armed with two or four U.S. built Sidewinder missiles on 30 pulsed 7.5 sec. before air-to-air rockets.

Final planned version, designated the J3F, will enter Sweden's air defense arsenal on a fully operational basis early in 1967. Already will be produced in greater numbers than one of the previous versions and production will extend through 1968. Prototype tests, using two converted J3Bs, began in 1965.

Primary difference in the J3F is the installation of an advanced version of the Saab 37 collision-course fire-control system, plus a more powerful reconnaissance/air control radar designed and built by L. M. Ericsson Co.

Felcons Used

Felcon will pack a one-two punch of four Felcon as its mainstay of two different types:

■ Felcon RB-327 (RB-35E) uses radar guidance and can be used as an all-weather but is limited at very low altitudes. It resembles the GAM-119 Navaho, and was specially designed to Swedish specifications. The high explosive warhead is detonated by a proximity fuse. Missing is an air-to-air battle (adjacent to the back in the aircraft's wing).

■ Felcon RB-328 (RB-35E) has a collision-course infrared homing system as

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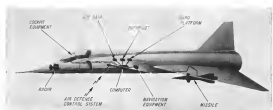
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MINIATURIZED, SOLID STATE computer core pilot workload on Swedish Air Force's Saab 37 Viggen. Computer location is indicated on the sketch showing its links with vital components.

expected to introduce 38, quadcopter. While aircraft restricted by rain and dense clouds, the missile can be launched at varying levels. Control line requiring a direct hit on the target is said. Cost of the missile is said to be considerably lower than that of the mid-grade Falcon, as the 38 usually probably would be used whenever conditions—such as very high altitudes, low airframe weight. It is carried on external wing pylons.

Both missiles are being produced under Hughes license by Saab in Lund.

Like the other Dornier versions, the 37F has advantages over the wing and is readily adaptable to the ground attack mission. Total armament in this configuration is either 12 Bofors 15.5-cm rockets, one 370F, or one 370F, one 1,000-lb. bomb, in addition to the aircraft's two 38-mm cannons.

During the two main aircraft exhibits, the Swedish Air Force considers its national aircraft more important than surface-to-air weapons. Nevertheless, a number of Bofors Bloodhound 1 missiles were bought three years ago for training and evaluation. They lived up to expectations and are now an operational asset. The second generation version, the Bloodhound 2, will be purchased this year for deployment against vital targets. Institutions are being established now, and the program is scheduled to be operational by the end of 1969. Bloodhound 2 quadcopter also will be employed against fast intruders at altitudes above 65,000 ft.

The air force has placed a particular emphasis on specifications during aircraft acceptance tests to be held later this year. Delivery is scheduled to begin in early 1965. The aircraft is slated to replace the de Havilland Ven-

om as an intermediate jet fighter.

The 38 can be quickly converted to a light ground attack aircraft by fitting it with external store mounts. In this mode it will be used to complement the strike force. The Viggen AS7 during the next decade. Targets will be divided according to capability. The 105 will attack enemy headquarters, while the 38, longer-range AS7 will blast the command fleet and stage the air to carrier taskings, at attacking supply lines, air bases, radar installations, communication centers and other tactical targets of a tactical mission.

A third is for the nuclear response, according to one military spokesman. Such weapons would be located in Sweden in conformity with Sweden's traditional defense strategy. Defense strategists believe that the air force is not directly involved in the probable, could be delivered by the Viggen.

Swedish development in this area would be hampered by the nuclear war factor. In the end, however, the decision will be a political one. The nuclear Swedish force is scheduled to determine its position on the issue next year. Conservatives have sided with the military on the question, but the Liberal Party is not yet decided. The expense of nuclear arms and the possibility of escalation if they were used in combat are likely to be of strong influence on the final decision.

At the present time the Swedish Air Force is organized into four operational aircraft groups.

• Group 1 is composed solely of strike aircraft divided into four attack wings of 12 squadrons.

• Group 2, entrusted with the defense of mainland Sweden, numbers three wings of four fighters and one of searchlight fighter wing, with 11 squadrons.

• Group 3 defends the central and eastern portions of the country with a mixed wing of one ground-to-air missile wing of two quadcopter and four A/W fighter wings of 12 squadrons.

This is the strongest air defense group in the air force, and its location is indicative of Sweden's concern over the threat to its industrial heartland from the Soviet-occupied countries of Eastern Europe and Estonia, located just 210 mi. away across the Baltic Sea.

• Group 4 anchors Sweden's northern defense perimeter with one wing of six fighters, one reconnaissance wing and one equipped reconnaissance A/W fighter wing, totaling nine squadrons.

In addition, the air force operates three main overhaul and repair bases and a flight training school, a bombing school, a radar and missile training school, an officer training school and school for training conduct in technical subjects.

Recent, the air force trend was toward a decreasing number of aircraft as the cost of modern airframe weapons systems soared. Five dayfighter wings (195 of the two) were merged into three from the winter between 1973 and 1980. These were mostly obsolete J29 Flying Banshees, of which 30 were sold to Austria.

With a total strength of 44 squadrons, each having 12 conventional aircraft, the Swedish Air Force today has approximately 530 combat-ready aircraft including those held in reserve and those in maintenance. This number is likely to increase as the old fighters at the day fighters are replaced by Dornier Replacement aircraft calls for the retirement of all 37F within one year.

Aircraft cost will see the phase out of the scanning Hawk Hunter squadrons, after which an A/W fighter interceptor units will consist only of 38.

whether Lancers and Dornier. Lancers will be the last to be replaced by the Dornier.

An issue's size of the current 5735 million defense budget is \$177 in billions, or about 30% of the GNP. That figure represents an increase of nearly 10% over last year. The cost is likely to become steeper as the new 1979-84 of a cost of about \$1.2 billion and the newer missile systems are purchased.

Total pay rise for Sweden 37 is reportedly about \$600 million, pointing to an area sharper upward trend in the 1970s.

In peacetime, the regular Swedish Air Force has a personnel quota of 1,850 commissioned officers, 8,400 war unit officers and 3,500 non-commissioned and related men. There also are 6,700 men performing support military services 190 WAFs and about 6,770 civilian employees.

The country's large number of non-commissioned officers is explained by the fact that all short-service pilots—those serving the minimum six years, which includes most combat pilots—are of that rank. These men could ultimately be made noncommissioned officers but as their policy dictates that officers be available in non-pilot situations.

Recruitment of more career officers, pilots and non-pilots in the short-service ranks is a drive.

Between 150 and 200 men, including both short service men and officers, are selected, are receiving pilot training annually.

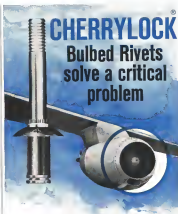
All pilots are trained for combat. At the air force's flight school in Ljungby, which they serve 20 hr of primary training in Saab 37, followed by 130 hr of jet instruction at de Havilland Vampire. After 15 to 20 hr of flight, they then proceed to the various combat wings or, if selected, to Dornier night training school. Withdrawals are said to account for more 50% of the men initially accepted in aviation cadets.

Inventory Reduction

Air Force Logistics Command reports an inventory reduction program including Project MINT (AFM No. 28, p. 101) have reduced the Air Force inventory by a net of \$74,000,000 despite the addition of about 400 new items daily.

For the first year of its operation, ended August 1961, Project MINT (Air Force Inventory Control and New Item Cost Method) is credited with reducing the inventory by \$74,000,000, or about 1% of a total of \$7,400,000,000.

The Air Force inventory had been reduced from an all-time high of 21,410,000 items in June 1961 to 1,867,000 items by August 1961.



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Eight Marks Claimed For Vostoks 5 and 6

Moscow-Soviet Union has announced it has filed claims for eight world records with the International Astronautical Federation as the results of the flights of Lt. Col. Valeriya Tereshkova in Vostok 5.

Rena claims two orbital class and two absolute records for Vostok's duration, 119 hr and distance 2,667,120 mi. World records for Mrs. Tereshkova are claimed for flight by a woman in duration, 71 hr, altitude, over 243,200 mi, distance, 1,223,055 mi, and payload in orbit, 10,149 lb. The class and absolute records are held by Russia, and the woman's class records set new.

An isolated decontamination system preventing the second class, Russia and the total thrust of the Vostok 5 and 6 launch systems was 1.2 million lb.

PRODUCTION BRIEFING

Thiokol Chemical Corp., Bristol Pa., has a \$17 million Air Force contract for qualification testing and production of motors for the Cassin-Argonne missile. Thiokol's previous Genac contract called for development and demonstration of a propellant motor. Genac's heritage operations and extreme operational temperature range.

East Sigbee, Inc., Santa Monica, Calif., has a \$1.6 million follow-on contract from McDonnell Aircraft Corp. for flight reference and bombing system of F-4C aircraft. The system provides the pilot with a visual display of his position relative to the earth and a series of commands necessary to execute the bombing maneuvers.

Bureau of Naval Weapons has announced its plans for 1971 will be additional Boeing Vertol CH-46A Sea Knight helicopters for the Marine Corps.

Marin, Calif.'s Delco Die has received a \$2.7 million Air Force contract for additional engineering services for the Pershing ground-to-ground mobile weapons vehicle.

General Dynamics/Pomona has been awarded a \$1 million Army contract for continued development of the B-600 medium-boar missile system. B-600 is a two-stage weapon against low flying targets.

Bell Helicopter Co., Ft. Worth, Tex., has received a \$2 million order for an extended number of Bell UH-1E as-

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such support helicopters for the Marine Corps. The order brings the total value of the Marine contract for the helicopters to over \$57 million since last January.

United Air Lines will pay \$17 million for Douglas DC-8 equipment south to Australia. Div. Aircraft Industries, Boston, Mass. The units will be used to equip United's DC-8s with four thrust engines for its new class aircraft. Seats will be equipped with passenger air and oxygen facilities, lights and a powered energy absorption system. DC-8s start in early 1964.

Bureau of Naval Weapons and Herndon Powder Co. have changed the management arrangement of the Navy-owned Allphase Scientific Laboratories, Cambridge, Md. All work since 1960 has been done under a single Navy contract. Government agencies and prime contractors desiring to begin product development programs at the laboratory will be contacted with the Navy. The new management allows other government agencies and prime contractors to deal directly with Herndon for work at the laboratory.

Instrument Div. of East Sigbee, Inc., Grand Rapids, Mich., has a \$170,000 Boeing contract to design, develop and build rate of climb indicator for the Air Force X-15 (Dyna-Soar) manned space plane.

AFOSR Awards

Grants for more than \$400,000 were given recently by the Air Force Office of Scientific Research to universities and research firms in the United States.

They were:

University of Texas, Austin—\$150,000 for work on aerodynamic and applied fluid mechanics.

University of California, San Diego—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, Berkeley—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, Los Angeles—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, San Diego—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, Berkeley—\$100,000 for research on aerodynamics and fluid mechanics.

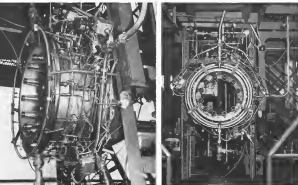
University of California, Los Angeles—\$100,000 for research on aerodynamics and fluid mechanics.

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University of California, Berkeley—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, Los Angeles—\$100,000 for research on aerodynamics and fluid mechanics.

University of California, San Diego—\$100,000 for research on aerodynamics and fluid mechanics.



TEST SETUP of 12 scale model engine to be used in a joint USAF-NASA program to determine the feasibility of augmenting a space booster's thrust through a shroud. Note access of test engine within perimeter of shroud. Tests will begin later this month.

NASA-USAF Test Thrust Augmentation

Huntsville, Ala.—National Aeronautics and Space Administration and Air Force will begin a joint experimental test program later this month to investigate the thrust augmentation provided a rocket booster by the addition of an air-scoop shroud around the base of the vehicle.

Needle is apparent to the shrouded rocket or ducted propeller aircraft the shroud is designed to increase the mass flow exhausted from the base of the vehicle. Air scooped in by the forward end of the shroud is mixed with the exhaust gases of the engines and then expanded and accelerated from the aft end of the shroud.

Theoretical Calculations

Theoretical calculations made by NASA's Marshall Space Flight Center, the USAF and the Martin Co.'s Denver Division, contractors for the test program, show that an optimum shroud design could increase the thrust of a vehicle by a factor of 1.5 to 2.5 times.

Where previous rocket engine shroud experiments have used night engines in industrial situations, NASA's USAF's approach is to use one large diameter shroud enclosing a cluster of engines.

This approach appears to promise good mass flow rates without resorting to extended length tubes in the duct.

In earlier shroud work, NASA personnel say, the shroud had to be so long to be effective that its weight negated any performance gain. Using multiple engines and a wide diameter shroud, the agencies believe that length-diameter ratios of 1 or less are not impossible.

Scaled-Down Engines

In this program, NASA and USAF are using 12 scaled-down engines in an initial configuration. That number was selected as the best quantity to produce accurate shroud flow rates through the shroud.

Each engine produces a 500 lb thrust and is a 1-1/8 scale model of the F-1 engine in the Space Shuttle P-1 engine in space operating characteristics, such as chamber pressure and area expansion ratio, based on smooth air-injection jets, the engines are pre-assembled every 30 deg apart along the outer wall surface of the shroud.

This is partly a research and development type of installation and was selected for ease of propellant handling during testing in wind tunnels.

The shroud extends to 16 in, the center body. The unit, with a conical aft end, resembles the nose section of a launch vehicle. Its aft end provides the effect of a plug engine on the gas flow in the lower region of the shroud.

Several shroud configurations will be tested. At least one will be convergent in shape while the others will be divergent. Reinforced aluminum structural shrouds to be used in this study test program will vary between 67 and 66 in in length and 32 to 36 in in dia. For length-diameter ratios between 1.2 and 1.3 shrouds will be tapered slightly, like truncated cones, and will have no fins at the base.

Inlet Geometry

Inlet geometries of the shrouds is fixed, since USAF and NASA did not want to complicate the program unnecessarily with variable inlet. Details of the air flow passed through the shrouds in the wind tunnels of USAF's Arnold Engineering Development Center, Tullahoma, Tenn., will be varied to maintain the net decreasing density of air encountered in an ascending trajectory.



AIR-SCOOPING SHROUD being tested by USAF and NASA scoop air at forward end. Air is mixed with engine exhaust gases and is expanded and accelerated from aft end.

Offered by Air-Scoop Shroud

One of the prime objectives of the 120 program now planned for the program will be the demonstration of an optimum shroud inlet area for a vehicle of given thrust operating between sea level and varying altitudes up to approximately 30 mi.

In operation, an air flow between the shroud and centerbody and be compressed to a pressure level slightly higher than that of the exhaust from the 12 engines. Flowing rearwardly past the engine ports, the air then would penetrate the larger volume of the lower half of the shroud, where it could mix with the exhaust gases of the shrouds, expand and be accelerated. Flow of the accelerated mass out the aft of the shroud should be at a velocity of about Mach 5.

Greater Mixing

To ensure greater mixing of air with exhaust, it is planned to install vortex generators toward the forward end of the shroud. Flow past the engines would be turbulent and, being somewhere from 2 to 5 psi higher than the ambient, would mix evenly throughout Marshall engines as they approach some secondary combustion is likely to

occur. Single injector of heat from the exhaust to the rocket-engine air should be sufficient to expand and accelerate the latter.

Rate of the augmenting mass flow to the exhaust flow is planned to be between 3-to-1 and 4-to-1, compared with the 10-to-1 ratios attempted in the past with single injector/single shroud tests.

The lower weight flow, say Marshall engineers, leaves the inlet requirements of the shroud and consequently reduces the drag of the duct.

The actual rates of tests at Tullahoma will depend on the weight and drag of the shroud, although it is clearly recognized that these factors could not approximately 20 to 30% from any performance increase obtained through the shroud.

Initial Tests

If the initial tests look encouraging, the agencies might experiment further with several refinements on the basic concept—including the addition of fuel injection in the forward end of the shroud, extension of the padded engines. Fuel would mix with the ambient air flow before diffusing with and burning with the hot exhaust gas exhaust of the engines. Although similar to an approach taken by Pratt & Whitney Aircraft in its research on high-pressure engines (AW Sept 25, p. 73), this technique would make the vehicle a true rocket-propelled, according to Marshall engineers.

Program was initiated by USAF's rocket test facility at Edwards AFB, Calif., last June. Marshall was invited to participate in the program, since it had available the facilities. Marshall's contract is for approximately \$100,000.

Soviet Article Derides U. S. Lunar Effort

Moscow—Comprehensive review of the U. S. lunar exploration program published by the Russian newspaper *Aviation i Kosmonavtika* (Aviation and Cosmonautics), an official Red Air Force organ, unambiguously criticizes the American effort. The article says:

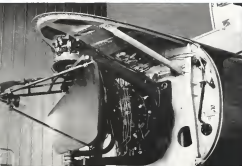
"From reports show that the American moon program is vulnerable in many points. First of all, existing and future (U. S.) lunar rovers have obviously made quite small steps."

"In long ago in April 1960 the Soviet Union achieved the power level to cover rockets within the United States is planning to reach in 1964-1967, after the Saturn 3 is placed in operation. Colors of existing and future space ships are not and do not indicate flight."

The program has been put together on the basis of the most optimistically stated possible for design and production of a huge number of very complex components. Neither U. S. scientific-research organizations nor the American aerospace industry has any experience at all regarding most of these components.

"The difficulty lies in building one of the subsystems will lead to disruption in the schedules for all related projects."

The Soviet report quoted Russia's space failures, including at least seven unmanned planetary missions and one unmanned lunar probe which were launched during the last 18 months.



Conceptual illustrations of: (Top left) Airborne surveillance-and-command system. (Top right) Integrated mapping, terrain-following, and airborne target-detection radar.

Left: B-58 bombing/navigation radar — an example of Raytheon's proven capability in the development and production of precision radars.

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detection radar for light attack aircraft — an airborne surveillance-and-command system featuring a long-range sensor to detect a variety of small targets "buried" in ground clutter — and a precision radar for accurate navigation, weather detection at jet altitudes, and approach monitoring.

For more information about these and other new Raytheon avionics developments, write: Nell A. Mostrom, Director of Marketing, Raytheon Company, Space and Information Systems Division, Bedford, Massachusetts.

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SOLID PROPELLANT PRODUCTION FACILITY-U.S. AIR FORCE PLANT #25 Lummus' architect/engineering design for this propellant complex included auxiliary facilities for MONUMMAN horizontal engine testing, propellant handling and storage, calibration, and rocket engine handling.



LUNAS ENVIRONMENTAL RESEARCH FACILITY-U.S. ARMY CORPS OF ENGINEERS Lummus performed the conceptual engineering for this simulator which established the engineering feasibility to provide for vacuum up to 7 x 10⁻⁴ torr, temperatures as low as -423°F, and solar radiation twice the intensity of the normally sun-up to 130 watts/sq ft.



WORLD'S LARGEST LIQUID HYDROGEN PRODUCTION FACILITY-U.S. AIR FORCE PLANT #74 Lummus designed, engineered, and constructed the hydrogen-production process plants of the plant to produce 70 x 10⁶ gal/year hydrogen, which is also liquefied through cryogenic processes.

PROCESS ENGINEERING MODEL, U.S. NAVY PROPELLANT PLANT, HYDROPLASTICIZER UNIT In addition to process design for neo-plasticizer unit, Lummus performed design functions for conversion of the unit from batch to continuous process, for a slurry-making plant to produce intermediates for possible fuel application, and for structural instrument-measurement facilities.



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and a study of tissue regeneration and wound healing during weightlessness—\$25,364.

• **Space Life-Dynamic monitoring of the cardiovascular system in weightlessness**—\$25,551.

Only 15 of the university contract grants have been fully negotiated up to now.

Some of the largest of these grants, the experiments and principal scientists are:

• **Monitoring bone functions and performance in the microgravity under prolonged weightlessness**—Dr. R. Adair, U. of Calif. at Los Angeles—\$92,194.

• **Passive hemodynamics and metabolic rates under microgravity**—Dr. U. of Calif. at Berkeley—\$90,188.

• **Study of pericardiovascular reactions under weightlessness**—H. Rudge, Jr. Wakefield, Johns Hopkins U.—\$80,000.

• **Study of renal and vascular changes produced by weightlessness**—Dr. D. Colquhoun, Macquarie U.—\$18,297.

• **Spontaneous and induced metabolic rhythms (Circadian) in isolated cancer as a temporal gauge of mammalian performance in extraterrestrial space**—P. Flaberg, U. of Minnesota—\$42,664.

• **Effect of weightlessness and other**

stressors on pathogenicity and immunity—Dr. T. Clark, Michigan State U.—\$43,991.

One additional experiment is for a study of the pathophysiological effects of weightlessness of patients with special attention to the role of the vestibular system.

A contract for this experiment is an extended \$300,000 to the U. S. Naval School of Aviation Medicine, U. S. Naval Medical Center.

General Electric Research Systems Dept. will develop and produce six flight models of the Biosatellite and a seventh for ground test. Its payload would range from 900 to 1,200 lb., depending on the duration of orbit. A Thun Drive launch vehicle is specified for the mission, but spacecraft weight may require either direct aspherization or use of a Thor Arrow.

Scientists, Employees Honored by NASA

Washington—U. S. space agency recently honored 23 officials for their contributions to the space program, including presentations for the first time of National Aeronautics and Space Administration Public Service Awards to two government scientists, and NASA gold medals for exceptional achievement to NASA employees.

New government scientists honored were Jack N. Jensen and Robert J. Parks of Jet Propulsion Laboratory for management of the Viking 1 Viking-80s program and John F. Yonkin of McDonnell Aircraft Corp. for major space program's Project Mercury activities at Cape Canaveral.

Gold medals for achievement were awarded to Dr. Elmer R. Chagnon of Ames Research Center for research in turbidity, energy physics and space electronics; Dr. Elmer Gosselin of Marshall Space Flight Center for his contributions to launch vehicle projects; Dr. John C. Marshall, formerly of Langley Research Center and now with Princeton Annular Research Associates, for developing the theory of the laser orbit rendezvous technique for the Apollo lunar landing; Charles J. Danilov, associate Langley director, for his part in organizing and developing concepts for the Mercury and Apollo projects and for mission selection; and for continuing successful space flight research technology, Dr. Walter H. Rousecroft of Marshall, for his work on launch vehicle and spacecraft systems, and Dr. William A. Munk of Marshall, for directing research and development in launch vehicle structures, mechanics, propellers and materials.

Another 14 NASA employees shared \$12,500 in cash awards for aerospace research.



REFRACTORY METALS APPLICATION NOTES

Do you know the facts -- or believe the fiction -- about processing molybdenum sheet?

Test yourself.

FICTION: A serious detriment to the use of molybdenum sheet is its "45° brittleness".

FACT: Recent advances in processing at General Electric have pointed the way to overcome this view of the problem. Actually, the greatest ductility of G-E molybdenum, as shown by tensile strength, is obtained by "tempering" in deep-draw cups, versus an "RT" or the rolling direction with a complete absence of brittleness under impact.

FICTION: Molybdenum purity is essential to gaining ductility in molybdenum sheet.

FACT: G-E molybdenum sheet is 99.95% pure. Tensile tests on G-E molybdenum show that the most superior ductility is quite comparable to higher purity in standard processing.

FICTION: Molybdenum sheet is "brittle" and will deformate in most bending or drawing operations.

FACT: Because of the fine-grained, cold-worked structure of optimum quality molybdenum sheet, there exists a natural tendency to deformation. In fact, even General Electric's advanced processing methods provide more sheet in thickness than the 1.0000" guaranteed no-deformation thickness in severe 90° and-reverse bend tests.



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AC-2 Flight Will Test Centaur Program

By Michael L. Yaffe

Cincinnati—Effectiveness of NASA's new Centaur program experimental will get its first test directly with the flight of the AC-2 development vehicle.

Some taking over Centaur project management from NASA's Marshall Space Flight Center one year ago, senators and congressmen here at NASA's Lewis Research Center have made considerable changes in the program. Most notable of these have been:

■ A five-fold increase in ground testing.

■ Simplifications in flight profile and overall mission requirements.

■ Initiation of studies aimed at minimizing payload complexity.

It will be some time before the effects of all the changes being made by Lewis can be fully assessed, but it is possible to review what has and has not been accomplished to date.

The greatly expanded test program, designed to catch most of the development problems on the ground, has led to a lot of late schedules and several changes in the vehicle. So far, the testing has not helped bring development costs or program slippage, but is expected to in the near future.

Flight of the second Centaur development vehicle, the AC-2 now at Cape Canaveral, is scheduled for next month (AW Oct. 14, p. 12). This is the fifth version as the launch date, which was originally in February, 1965.

Cost Estimates Double
Total development cost of the Centaur program, based on the present eight-vehicle schedule, is now expected to run close to \$720 million—about double estimates made in the middle of 1961 for a 10-vehicle program.

By the end of Fiscal Year 1965, the government had spent approximately \$120 million on Centaur development. For Fiscal Year 1964, NASA has set \$112 million and expects to spend an other \$45 million before development is completed in 1965.

On the other hand, the expanded ground test program has produced several changes that are expected to increase vehicle reliability. Among the most important of these are changes in the second stage start sequence of the Centaur.

These changes, all checked out in an extensive series of ground tests, are expected to provide more reliable loading.

■ Faster and more reliable stage separation.

■ Some increase in payload.

■ A significant shortening of the time

during which the vehicle is subjected to the unknowns of zero-g flight.

The new stage separation system, which will be tested on the AC-2, uses a laser shaped charge to cut the welding band joining the Centaur to the interstage adapter and eight retro-rockets to lock off the empty Atlas and allow the Centaur stage to pull free of the 134 adapter section.

On the old AC-1 (then called R-1) flight in May, 1962, pneumatically actuated latches joined the Centaur to the interstage adapter, and two 50-lb.-thrust pyrotechnic ejection rockets were used to push the fully loaded Centaur out of the adapter section.

Old System

In the old stage separation sequence, the Centaur took about 12 sec. to pull out of the interstage section and, with the latch system, was more likely to become against the adapter. In the new separation system, the Centaur takes only 2 sec. to clear the interstage adapter.

In a recent test at Lewis it managed a clean separation under simulated adverse conditions caused by the simulated loading of one of the eight solid propellant retro-rockets jiggling the aft end of the Atlas. The laser shaped

charge also weighs approximately 25% less than the pneumatically actuated latch arrangement.

Another important change—also to be flight tested on the AC-2—is the first time a ground checkout of the RL10 for hydrogen engines. This technique was checked out in ground tests at Pratt & Whitney's Florida R&D center. Liquid helium is used to cool the RL10 engines to 100 R (-197.7 F) while they are still on the ground.

Airborne Checkout

During the first run of boosted flight, the second stage propellant system temperature was to 100 R. The system then undergoes an abbreviated airborne checkout to 150 R to prevent the propellant from freezing into vapor, and the engines are started. Liquid hydrogen from the second stage fuel tank is used for the airborne checkout and then is vented outboard.

From stage separation to second stage ignition, the new sequence is approximately figured to take 15 to 16 sec.—5 sec. for firing the Atlas motor engines to stabilize the vehicle after first stage burnout, 2 to 3 sec. for Centaur to clear the interstage structure, and 3 sec. for Airborne checkout. Lewis engineers hope to shorten this time by 5 sec. or



SKETCHY ATLAS-CENTAUR in its stage separation tests at Lewis Research Center is shown below separation. After retro-rockets are fired, hooks on the aft end of the Atlas, which is suspended from the overhead hoist, engage the net and keep the Atlas from bouncing back into the mock Centaur stage which remains stationary.



Four-Stage Athena Mockup

Athena four-stage, solid-propellant rocket, being developed by Altkor Research Corp. for the Air Force's advanced ballistic re-entry system (ABRES) program (AW Sept. 2 p. 36), is shown in mockup form. Plans are to launch 77 of these vehicles with payloads consisting of a variety of exotic bombs and precision strike. Plans are to launch payloads weighing between 50 and 300 lb. at altitudes up to 100,000 ft., then bring them back through the atmosphere.

possibly more by eliminating certain engine stabilization and by reducing the time which has been required for air-borne children.

In comparison, the original Soviet stage start sequence in which the chaff down was carried out completely in the air using induced propellant required 45 sec—51 sec for airborne children,

and 5 sec for greater stabilization and 7 sec for clearance of interstage structure.

Among other things, this meant that more propellant had to be used for cooling and three engines cooled, and that more propellant had to be used to make up the engine lost in the vehicle during this longer period of coasting against gravity.

Simplification of the Centaur flight profile to a one-burn direct ascent to the moon is another important mission that must be done.

Early plans for a three-burn, parking orbit approach were later modified by Marshall Space Flight Center to a two-burn, parking orbit flight.

By going to a one-burn profile, Lewis

hopes to obtain a slight increase in payload capability and greater flight reliability by avoiding the problems associated with parking orbits such as the effects of aerodynamic heating.

However, since the one-burn, direct ascent profile requires the launch vehicle, particularly during the ascent, Lewis acknowledges an increase in the possibility of retaining to NSRCC's two-burn program in later development flights.

By the same vein, overall mission requirements have been simplified for the Centaur development program. Centaur no longer has to be able to stage to all payloads, says NASA's Vincent E. Johnson, Centaur program manager.

Payloads scheduled to hitchhike on Centaur development flights have been transferred to other vehicles thereby making it possible to hold down costs and, more important, to permit payload designers to concentrate on a Centaur itself during this critical development period.

Because of the changes in the Centaur flight profile and in the second stage starting sequence, Lewis says he now believes that they no longer will need the four-period alloy rockets designed to keep the propellant tankage space on top and laid in the original separation procedure.

Alloy racks in the proposed two-burn flights will be avoided from their present position toward the low end of the aft end of the Centaur stage between the RL10 rocket nozzles.

In a one-burn flight profile, Centaur would probably park in orbit for 70 to 25 min. Although the exact parking time hasn't yet been established, this period—whatever its duration—would be expected to prove a critical one.

To find out what might happen to the vehicle at this time, engineers at Lewis are modifying part of the Space Power Pylon into a virtual space simulator. A complete Centaur second stage will be tested in it for stability at 10,000 and 100° sec. It also will be exposed to simulated solar radiation.

Practically, the Centaur stage will be modified to incorporate new flight configurations. It will undergo a new-kind checkout of its electronics by dynamic, telemetric, attitude control and guidance systems.

Lewis engineers strengthened the retesting adapter because tests indicated high potential wind loads and because there are more failures in the adapter section now. Among other things, they increased the number of longitudinal struts in the retesting structure. Weight increased, but because the adapter section starts with

the Atlas there will be only a relatively low payload reduction.

Wall thickness of the liquid hydrogen tank has been increased from 0.010 in. to 0.015 in. on the AC-1, to make possible better and more reliable seals. But Lewis engineers consider this as only a temporary solution and are investigating the use of chemical welding and improved welding techniques as possible alternatives. They also are considering venting to a material strength than the stainless steel that is currently being used.

Another weld problem was Lewis engineers hope to correct on the fly. It is the lower-level connecting the interstage adapter, side porch, all bulkhead connections and the interstage bulkhead. This is a complicated welding spot which they plan to simplify and strengthen after the AC-2 flight.

The redesign work done on the AC-2 has been redesigned as the result of a series of wind tunnel tests to determine what happens in the hydrogen phase under various conditions of velocity.

It will extend further out from the stage than in the AC-1 to keep it away from the boundary layer and the main engines, and prevent the possibility of the phase shifting. Looking back on the real tests, which are being

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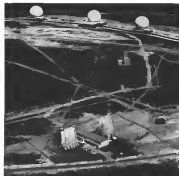
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Fylingdales Moor Radar Site Shown

Aerial view shows the one radar warning site at Fylingdales Moor, Yorkshire, England, which went into operation recently. Three others face north and east to provide backup of their signals over the North Sea. Triangular radar communication facilities in three is largest.

clouds during critical parts of the flight, have been changed for the AC-2 flight.

Such baffles similar to those used in the AC-1 have been added to the Centaur stage for tests. Dynamic ground studies have shown that the forces produced by the shaking of the Centaur's more than 20,000 lb. of fuel are more critical than was first anticipated and could become high enough to distort the vehicle. For later vehicles Lewis is considering lowering the low tank ballast to reduce the available shock.

Many changes have been made in the launch ballast, once thought to be the source of failure in the last Centaur flight. The design incorporating the new core to the forward ballast has been strengthened and at the same time, modified to reduce the support loads on the stage.

Lewis has redesigned the motor shield and profile reinforced the whole area. But, with all these tests and studies, engineers at Lewis still don't know exactly what caused AC-1 to fail.

The large vibration panels around the liquid hydrogen tanks have been changed on the AC-2 and will be further modified on the AC-3 to improve their structural integrity.

Lewis engineers are changing the thickness of the adhesive used to bond the glass fiber insulation. Steel bolts which hold the heavy grade on the AC-1 have been replaced with more positive flange type fasteners which will be cut into by a shaped shape to sit on the panels. In the AC-2 flight, however, the membrane will be bolted on and not jetted.

The guidance system on the Centaur will remain the same, at least for the time being. Lewis project engineers decided that it was the best available but required more testing and improved quality control procedures to ensure reliability. Studies now under way are aimed at further improving the all-in-one guidance system.

In addition to the extensive ground test program and the actual change in children procedures, a noteworthy acquisition development has been the issuance of contracts to study the vehicle's performance and payload capabilities, which have been falling off from the last design figures during the course of the development program (AWF Oct 14, p. 22).

The studies might have a strong influence not only on the Centaur program's future, but on later rockets.

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rocket engine projects and on civilian aerospace applications in general.

Last April, Lewis awarded the Martin Co.'s Denver Div. a \$335,000 contract to evaluate the Titan 2 as a booster for Centaur.

In July, NASA issued a \$980,430 contract to Pratt & Whitney for research on the use of Ecuarc and fluorine-oxygen (fluor) motors as outflow with by design, using components of the RL40 rocket engine which P&W is selling for Centaur.

Two weeks ago, NASA declared that it had awarded \$566,991 to Rockwell and \$209,673 to General Dynamics Aerospace to study the use of how in the Atlas program version.

According to preliminary studies, this is how the propulsion alternatives meet or not as a performance basis.

• The two-stage Titan 2 would provide a 25-10% payload increase over the



Dyna-Soar Suit

USAF/Borg N70 (Dyna-Soar) suit prototype already pilot test protection through the spaceplane test program. The suit is being developed by the Air Force Research Command's Aerospace Research Div. Suit will be tested in a series of 11 live jumps (X-45 Sept. 30 p. 17) at El Centro. Cold Weather test on May 31 W. Wood. Dyna-Soar test pilot.

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present now and a full stage Atlas.

• Performance of an Atlas-Centaur using four in the Atlas booster would be slightly under that of a Titan 2 and, even less in the second stage Centaur as well, slightly above that of a Titan 2.

• Flow-burning Atlas combined with a Centaur second stage burning straight flameless oxidizer would be even better.

Johnson emphasizes that these alternative propulsion possibilities are simply being studied now and are not under active development.

At this point, he says, it is still too early for any definite decision. A decision to use one of these alternative propulsion schemes in the Centaur will depend upon such factors as future mission requirements, cost comparison of the alternatives and availability.

Other changes in the Centaur program by Lewis include redesign of the nose fitting, improvement of fabrication techniques, dynamic testing and complete structural analysis of the nose Atlas-Centaur vehicle.

In some cases, such as in some of the propulsion and guidance systems, Lewis has expanded and improved programs already under way when it took over management of the Centaur program.

These changes in the Centaur program, particularly the great expansion in ground testing, came about for several reasons.

One was the skepticism by Lewis engineers and scientists, underscored by investigation of the earlier flight failure of the AC-1 test vehicle, that there was still a great many things they didn't know about the vehicle.

Another was the philosophy of Lewis, as expressed by the center's director Dr. Alec Stevenson.

"Everything that can be done on the ground in the way of test and checkout must be done perfectly on the ground, and in this way I think we can find 95% of the problems."

Also important, of course, was the fact that more money was put into the program to make this increased testing possible.

While the expanded test program has added to Centaur development costs, a Lewis spokesman says every effort is being made here and at other government and contractor test sites to hold these costs to a minimum.

At Lewis, for example, Centaur project engineers have used existing test facilities, which have had to be modified extensively in some cases, rather than build a completely new test facility for the program.

Of the new Centaur test facilities now being built, the largest and most expensive is the 56-ft-dia test chamber



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where test stand now being built at San Diego.

Scheduled for completion late next year, the stand will be owned by NASA and operated by General Dynamics/Astronautics.

In it, Atlas, Centaur and the Saturner payloads will be tested thoroughly for an end-to-end, horizontal element of all-volcanic engine tests involved in actual space operation.

It is expected that the stand will shorten the time required for testing at Cape Canaveral.

Until the stand is finished, NASA plans to use CD's reference test facility, which cannot handle the Saturner payload.

The new vertical separation test facility at Point Loma, Calif., which is much smaller, was nearly completed when Centaur took over the program, and it will be leveled off even though it offered more limited stage clearance space than Centaur's own stage separation test facility at Glenwood.

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APOLLO TELECOMMUNICATIONS SYSTEM must provide wide variety of services, including error data and telemetry transmission from Lunar Excursion Module and command module to earth, as well as voice-data communication between the LEM and command module and radio intercommunication between astronauts on moon's surface. Location of telecommunication equipment in the command module is shown in detail, right.

Apollo Communications Details Disclosed

By Philip J. Kloss

Mission Brief—First details of the Apollo telecommunication system, the most complex spacecraft radio-telecommunication system yet developed, were reported here during the recent National Space Electronics Symposium.

The increased complexity results from the variety of communication functions which the system must provide. These include astronaut-to-astronaut within the spacecraft, astronaut-to-earth, command module to Lunar Excursion Module (LEM) during and after the lunar landing, ranging astronaut to LEM to command module or to earth stations and ranging astronaut to roving astronaut.

The report on the Apollo system was made here in two papers. One was authored by Samuel W. Foxworth of the National Aeronautics and Space Administration, and the other was jointly authored and delivered by W. S. Pope of North American Aviation and Dr. S. E. Wilmore of Collins Radio Co. Collins is the prime contractor for the Apollo telecommunication subsystems.

Distance Equipment

The Apollo mission, according to Pope, will spend communication equipment across a 210,000 mi. path—descending ascent in the Atlantic Ocean, ascending orbit in the ocean, leaving some to orbit the moon, leaving some to a space orbit during re-entry, and returning a little to earth in the location and accuracy of the approach since in flight has ended.

The VHF (very high frequency) pattern of the spectrum used in Project

mission and in the LEM. This two-station-to-two-station system is a radio transparency for internal tracks and across to transmit and receive voice, data or TV. The portion of the Apollo telecommunication system therefore is referred to as the "linked 3-band tele-system," to denote its multiple function nature.

Initial tracking of the Apollo vehicle during launch, re-entry orbit and trans-lunar injection will be accomplished by NASA's and USAF's existing network of AN/VTS-94 and AN/FPS-6 C-band radars. Separate C-band (Henderson) will be entered in the instrument net (located between 5-18 stage and separately) and in the (initial) command module.

Tracking Functions

Power plans call for the C-band radar to handle the tracking function until the approach has reached an altitude of several thousand miles during the re-entry phase, after which NASA's deep space tracking stations in California, South Africa and Australia will take over, tracking on S-band.

These three stations can maintain the Apollo approach in continuous view only when it has reached an altitude of nearly 10,000 mi. or, possibly beyond the effective range of the ground C-band radar. However, in holding one "top-of-the" S-band radar at key points, it may be possible to begin a

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band tracking at altitudes of about 2,000 feet or, providing sufficient control with the C-band link.

During initial staging of the Apollo launch vehicle, as it is brought to a circular parking orbit at an altitude of about 110 nmi, a decision must be reached as to whether all elements are operational and whether or not to acquire the S-band range for telemetry support. Data on the condition of the spacecraft and the astronauts will be received down by means of a pulse-code modulation/frequency modulation (PCM-FM) telemetry transmitter operating in the 125-200 mc band at a rate of 31.2 kilobits per second. The

VHF FM transmitter will have an on-off power level of 10 w. Postorbital test the instant response. The antenna will be capable of following up to 270 different functions in true response. Telemetry data will include both analog and digital information, which will be relayed to the NASA integrated mission control center (IMCC) at Houston.

On board the spacecraft, the three astronauts will communicate with one another by means of an intercom net. A control panel for each astronaut will provide flexibility so that two astronauts can talk to each other while the third is talking to earth. Intercom

connections also are provided with the LEM for astronaut communications during the lunar-orbit phase of the mission.

During the near-earth orbit phase of the mission, data will be transmitted up to the spacecraft from lunar-orbit altitudes at a UHF frequency of 400-450 mc. This will include ground-to-orbit data as spacecraft pointers and vehicle status information with analog data obtained from a vehicle's self-contained guidance and navigation system (AGN) Sept. 30, p. 32). Also transmitted on the "uplink" will be vehicle status information commands issued by the IMCC. The latter can be overlaid by the astronauts by means of switches. Postorbital

Trace-Laser Injection

When the decision is made to proceed with trace-laser injection, the S-band range will be acquired to give an additional velocity component of about 16,000 ft/sec. When the spacecraft reaches an altitude of several thousand miles where C-band tracking is at the limit of their range, the injection should be well established. Postorbital test the instant the spacecraft's VHF AM voice signal would be of acceptable strength for reception by tracking facilities, and telemetry data would have been lost.

One possible solution is to exploit the present quadrifilar ground system with 60 ft dikes, providing an additional 14-db gain, corresponding to a fivefold increase in usable signal. Additionally, the data rate for telemetry would be reduced from 31.2 to 3.6 kilobits. An alternate proposal under consideration is to switch from VHF to the unified S-band transmitter for telemetry communications at an altitude of about 1,500-2,000 feet, or, for this poor problem, become a more complex high tracking rate for ground stations and provide such short tracking periods. Present plans call for further tests on the latter approach during the early Apollo qualification tests. Postorbital test.

S-Band System

Once the switch-over is made from VHF to the unified S-band system, the NASA deep-space communication facility (DSF) operated by the prime contractor will receive the status of voice and data communications with the spacecraft. Voice and data sent to the spacecraft will be transmitted at 1,180-1,190 mc while the spacecraft will reply at 2,340-2,350 mc. When operated as a tracking orbit, the DSF station will transmit a pseudo-random noise code which will be transmitted back to the phase-locked spacecraft S-band transponder. From the returned signal, the transmitted

time is expected to be able to measure spacecraft distance (range) to within 10 ft. Postorbital test. The range-measuring technique is expected to be sufficiently accurate to permit determination of range-rate as well as distance, enabling NASA to compute spacecraft position and provide position/velocity data of the astronauts can compare with data obtained from the artificial path and navigation system.

These components of spacecraft position and velocity will be made available during the passage of the moon and at several three-point to four-second intervals in place the spacecraft is a lunar orbit, roughly at an altitude of about 30 nmi.

While the spacecraft is in lunar orbit, the moon will shield it from direct sunlight communications for about half the orbit, or roughly 15 min. During blackout, telemetry data will be stored in a tape recorder for subsequent transmission as appropriate data when the vehicle comes within view of the earth.

Descent Communications

When the LEM separates from the command module and begins its descent to the moon's surface, communications between the two spacecraft will utilize the VHF-AM between the receiver originally employed for command module-to-earth communications when it was in near-earth orbit. Voice communications from the LEM astronauts will be transmitted direct to earth from its own S-band transponder, or can be relayed via the command module through the VHF-AM link if any reason the earth station can not get through to the LEM while it is in transit.

Once the LEM has landed on the moon, the astronauts will erect a directional antenna about 10 ft in diameter. The antenna gain must be increased and the antenna must be oriented so that the antenna is directed to transmit station on its unified S-band antenna. During voice-data transmissions, a power output of 5 w will be used.

The power output will be increased to 20 w to transmit TV and higher-power broadband type data.

When one of the communications antennas from the LEM for exploration, a dual VHF-AM belt-pair transmitter will provide radio communications to the other astronaut within the lunar spacecraft which in turn will be relayed by VHF-AM to the final antenna in the command module, when it is within line-of-sight range. The portable transceiver is expected to provide usable range of up to 9 ft.

Television pictures of the lunar surface will be obtained using an existing type TV camera with a monochromator's focused vision tube. It will have two lenses, one 18 in. in diameter lens providing a field of 9.55 deg

and a wide-angle 9 mm. lens providing a 68-deg. field of view. Camera will have an automatic sensitivity control to adjust exposure to ambient lighting conditions.

A 130-lb. unit will be employed with a frame rate of 12.5 per second, providing a picture slightly narrower than that of the astronaut. The camera will weigh about 5.5 lb., consume about 6 w., Calfax Radio's Weston reported.

When the LEM takes off from the moon for orbital rendezvous with the command module, both spacecraft will be within view simultaneously of an earth tracking station permitting it to

monitor the initial phase of the operation and provide voice assistance. But optical and radar devices on board the spacecraft will provide the precise position information needed for lunar rendezvous. During the rendezvous operation, communications between the two spacecraft will be provided by their respective VHF-AM transmitter-receivers.

Following crew transfer to the command module, the ascent module engine will be fired to inject the spacecraft into the transfer orbit back to earth. Voice and data communications will be handled by the unified S-band

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• **Measuring Hypersonic Plasma With Laser—General Electric** is about to receive a contract from the National Aeronautics and Space Administration to investigate the use of a laser laser for measuring the hypersonic flow field plasma parameters from the molecular scattering of light.

• **Unfolding Solar Collector to Get Space Test—A** space test of an experimental 10-ft-dia. unfolding solar collector which could gather the sun's rays and beam them onto an energy conversion device will be conducted by Lockheed Martin & Space Co. aboard an Agena spacecraft launched in a folded configuration, the collector will unfold in space much like the petals of a rose. The collector will be built by Electro-Direct Systems under a program called SORUS (solar orbital receiver system).

• **Interaction of Laser Beams With Matter—Duke** is investigating into the effects of laser beams on optical components. In high power laser systems will be conducted soon under a forthcoming Army Signal Corps program. The investigation is aimed at answering proposed optical components and will cover on multi-layered dielectric reflective coatings, anti-reflection coatings, lenses, prisms, and the laser radiation itself.

• **Spayed Solar Cells—Research** on chemically sprayed thin film solar cells is being conducted by the Physical Research Dept. of National Cash Register.

• **Signed on the Dashed Line—Major contract awards** recently announced by aircraft manufacturers include the following:
• **Electronic Communications, Inc., Research Div., Tuscon, Md.,** will develop a multi-segment adaptive systems system for removing telemetry data from distant space vehicles, under \$53,000 contract from NASA's Goddard Space Flight Center.

• **Honeywell's Aerospace Div., Minneapolis,** will prepare preliminary designs for high performance visual system components suitable for Nike X missile under letter contract awarded by Martin Marietta.

• **Radio Corp. of America** will study variety of all-weather landing systems for Army aircraft under contract for \$200,000 from Army Electronic Research and Development Laboratory. The 12-month study will be conducted by RCA's Aerospace Communications and Controls Div., Camden, N.J.

• **Raytheon Corp., Canton, Mass.,** has

awarded contract for \$82.6 million for 17 additional 10-15 days predicting systems to be used with USAF's B-57D Intruder Control (B-57C) program. This extension covers total contract price to \$37 million for total of 14 systems.

• **Long-Term-Vought** will build and install an Inertial Reference System for NASA's Nimbus-Terra meteorological satellite control center at Goddard Space Flight Center under \$45,000 contract.

• **Electro-Mechanical Research, Inc., Seattle, Wash.,** will design and build machine simulation system. Industry equipment for Army's Heavy Demolition Laboratory and for Sandia Corp. will do two separate contracts. Army contract calls for a voltage-controlled oscillator whose operation does not deteriorate under a peak power flow of 100 megawatts/sec and no integrated system flux of 10° maintaining on having energy greater than 10 J. The Sandia contract calls for development of electronic systems, low level amplifiers, voltage regulator, commutator and calculator.

• **Space Phoenix Co.** will continue its efforts to define control display system integration and pilot functions during approach and landing of advanced aircraft under USAF's Aerospace Systems Div.

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envelope which can be bonded to curved or flat surfaces. Manufacturer: Tylco Corp., 4335 Spencer St., The Wood, Calif.

• **Cutters** use sharp, tungsten carbide bits to cut, grind, and drill. They are used in the production of dies and molds.



with made of tungsten, phosphor bronze and beryllium copper. Design, intended for laboratory use, has pin vice and drive mechanism for holding. Manufacturer: BCI Research, 1518 York Road, Timonium, Md.

• **Radio receiver** with continuous right circularly polarized elements, Model 30-40, for operation in the 2-4 GHz (dec) band, measures 100x100x100 mm and weighs only 40 lb. The unit, suitable for electronic warfare and plasma wave



radio applications, provides eight simultaneous overlapping independent beams with uniform taper across beam with cosine taper or six beams with cosine-squared taper by use of beam combining circuitry. Manufacturer: Advanced Development Laboratories, Inc., 24 Reed St., Norwalk, Conn.

• **Minimotor** is a motor without, Magna Series II, measures 1 in. in diameter and 1 in. long, produces 1/20 in. torque and consumes 0.1 watts power. Solenoid complex balanced construction with no axial movement of shaft end is available in variety of sizes. Manufacturer: Magna Inc., 810 No. Wacker Drive, Chicago, 60606. (Continued on p. 102)



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And, these programs are diverse. For example, plans are under way for:

- simultaneous range support for time-critical complex missions.
- increasing the accuracy of pulse and CW radar systems through ultra-pressure and near-real-time calibration using satellites.
- the use of instrumentation facilities to escape from the effects of the earth and its atmosphere.
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LETTERS

West Ford

I have just read your article on the West Ford experiment in today's (Sept. 30, p. 88) issue of *Astronomical Watch* & Space Technology. Early paragraphs in the article read as follows:

Peak transmission rates were obtained only during the early days which followed formation of the coherent double helix.

The white spot, belt of copper mottles formed completely about June 15, 1 day after the dispersing mechanism ceased flow.

*Transmittance of information has now decreased to about 1,000 bits/sec.

It is not clear to me how this supports whether you can argue that the "post-immunization effect" was in fact measured in the first week after the diphtheria was ignored by the deponent, when all the diphtheria were still concentrated along only a few thousand miles of the 40,000 mi. or so coast, and long before they were disbursed all around the circle to form a "belt" (see diagram).



On June 18, the code of the long thin circles cloud of dipole first began to evolve by and increasing, and the distribution of dipole around the circle belt became increasingly uniform for several months thereafter. The data set of 2 000 lat/sec refers to the performance of the completed belt in the latter part of August, when the belt had "flashed completely."

In view of your continuing interest in the project, I want to let you that this sequence of events is clearly established.

Joan A. Kozmin
Massachusetts Institute of Technology
Lincoln Laboratory
Lexington, Mass

(**IGNITE** WEEK & SPACE TECHNOLOGY) used the description of Walter E. Mosley Jr., of *Lancaster* Laboratory, cited in the story, and that, "higher temperatures than were achieved only during the rush down the hill." Since the use of the term led could imply a full engine, Mr. Kasher's letter renders any ambiguity that might have arisen. —E.J.)

Cooper (Cont.)

With regard to reader Stranberg's letter (JAW Sept. 58, p. 106), I should like to state that, contrary to the inference, astronauts are acutely aware of the effects of atmospheric situations. The geometry proposed by Mr. Stranberg is essentially correct in describing the magnifying effect of the

Attention: *Work* welcomes the comments of its readers on the issues raised in the magazine's editorial columns. Address letters to the *Editor, American Work*, 238 W. 42nd St., New York 36, N. Y. Try to keep letters under 300 words and give a precise identification. We will not print anonymous letters, but names of writers will be withheld on request.

discovery. His explanation would have been more likely, however, if he had not been so sure of the correctness of his method of selection of an α in each β apparent match. 1,000's and we may take the classical etheric scale length to be roughly 5. α 's. Thus to see an electron in space the atmosphere would make objects in the earth's medium appear 5 ft. closer—nothing so serious as their apparent dimensions of 0.0019% is seen from an orbital altitude of 100 mi. This should conservatively stretch out our earlier conclusion that the atmosphere plays a negligible role in the apparent size of objects. A more serious world have been better taken and Wile Cooper would the earth's surface through 190 mi. of water.

BRANDON A. SWEN
Associate Administrator
New Mexico State University
University Park, N. Mex.

Not to believe the point, but if the "Mrs. Cox" episode can be seen in what he observes in north-Texas, nothing by one of the Italian Ocean ship of Schenck—who should we doubt an intelligent soldier to see some possible obscure no. 1000?

That must prove the question Cooper is asking is clearly these subjects is to better know his capability to make intelligent conclusions.

WYNELL SYMONDS
Lancaster, Calif.

'Big Lift' Useless

I have been watching, with increasing dismay, the science being played on bagpipe transport around the corners of troops in potential battlepositions around the world. Operations of this type are only feasible during peacetime, while 100000+ casualties are in perfect condition. Loading and unloading are inoperative and support equipment is readily available. In the context, the proposed Exercise Bag Left is completely senseless in that it is possible movement of troops is, in fact, impossible.

As long as military transport must all be prepared to move, more detailed and detailed to be short in numbers and size the whole concept of such troop movements is not practicable in wartime.

The Soviet Union wants to be on the right track with this project to build 2,000-passenger, water-based aircraft. It may not be feasible with the present knowledge to manufacture an aircraft of this size by the advantages of operating from water as shown, even though previous operations

costs would be much higher, due to the unfavorable payload/weight ratio inherent in such aircraft.

In view of the above, the idea of reducing troops stationed in Europe by one division each year, based on the availability of reliable military air transports, is rejected. People who should know better, such as Sen. McNamara, will have to take a great deal of blame for our disastrous consequences, such as a costly war in Laos.

B. SUNDHANTEN
Stagefitter Enterprises
New Orleans, La.

Credit Given

Your story on our Cavalier (AOL Sept. 5, p. 180) was, as usual, a good one. I would like, however, to add an omission on our part: the Access Lab of the New Party's AccessSpace Systems Division should have been credited with support of the program through its conception and development (10/93).

H. SHILLMAN
Manager
Information Systems
General Postcard, Ltd.
Aeroplane Garage
Little Falls, N.Y.

Electron Beams

I would like to point out that in the article in your Sept. 15 (p. 38) issue entitled "Euthydia Farm Techniques Make Impact"

Scanning Electron Microscopy,¹ a significant advancement was made in the exportable file of the Dr. G. G. Wells in the work described. In fact, the Scanning Electron Microscopy was being used by three laboratories was designed and constructed primarily by Dr. Wells, and he has also carried out many of the investigations either in person or by others.

I. M. MacKenzie, Manager
Information Systems Dept.
Research & Development Center
Westinghouse Electric Corp.
Pittsburgh, Pa.

Argosy Error

Re "Agony Production" (AM Sept 3, p. 25): As an Slicker should have no trouble selling the 10 Screen 280 Agonyes, imagine a 90,000-lb. airplane with a 300,000-lb. payload! Did an extra zero creep in unnoticed?

Seriously, all survivors said: you are to be congratulated on a fine magazine, which has in the last three years, given at least one former engineer a most informative and occasionally inspiring view of his future in the

MICHAEL J. GAYNE
Cornell University
Ithaca, N. Y.

(Our website The Agency will carry \$1,000 B. period—\$4)

Today... Bol's Agena rocket engine leads the way in space Agena has orbited more than 65 percent of all USAF and NASA satellites and has placed 80 percent of the free world's unrimmed payload in space. Agena placed the Ranger 4 space craft on the moon and powered the Mariner 2 on its historic Venus fly-by. Current assignments for Agena include OGO, POGO, Aera, Nimbus, Echo I, Gemini Target and Mariner Moon plus many USAF missions.



1985 — the as a first hand and judge networking for dynamic system engine :
 successfully run first by staff



Tomorrow... Bell's high energy rocket engines will set the pace High energy pro-

peloid work was initiated at Bell in 1948 with test firings as a chlorine trifluoride/hydroxide rocket. Liquid effluents began in 1956. Bell achievements under USAF and NASA contracts in the fluorine propulsion field include hundreds of firings at thrusts up to 35,000 pounds as well as the handling and consumption of over 100,000 pounds of liquid fluorine with a perfect safety record. This effort has resulted in the resolution of all major fluorine propellant problems. The fluorine propulsion experience of the current Bell staff exceeds 300 man-years.

Bell's work with high energy propellants continues at a high level . . . ready for tomorrow's space needs.



BELL AEROSYSTEMS COMPANY

OCT. 23 1963

"AND FACTOR"

- The Phantom II can operate with combat loads,
from runways less than 5,000 feet,
- AND...** deliver a conventional bombload of 4,500 pounds,
- AND...** utilizing one of two Bullpup guided missiles,
visually sight and destroy a railroad bridge,
- AND...** with the other, wipe out a missile launch hard-site facility,
- AND...** with the still remaining Sparrow III missiles, destroy enemy
aircraft or air-breathing missiles, even in head-on attacks,
- AND...** operate in any kind of weather, day or night,
- AND...** accomplish all these missions 500 statute miles from this base,
with fuel left for the return.



Defense planners recognized this "AND FACTOR" versatility when they chose the multiple-mission Phantom II for the United States Navy, United States Marines and United States Air Force.

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